

DIGITAL SIMULATION AS AN APPROACH
TO ESM SOFTWARE DEVELOPMENT

Robert Daniel Stanga

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THESIS

DIGITAL SIMULATION AS AN APPROACH
TO ESM SOFTWARE DEVELOPMENT

by

Robert Daniel Stanga, Jr.

March 1975

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Digital Simulation as an Approach
to ESM Software Development

by

Robert Daniel Stanga, Jr.
Lieutenant, United States Navy
B.S.A.E., Auburn University, 1968

Submitted in partial fulfillment of the
requirements for the degree of

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ABSTRACT

Selective problem areas relating to computer based ESM system software development are discussed. Digital simulation is presented as a practical approach to solving software development problems. An algorithm to simulate a hard-wired EW signal preprocessor is developed. An application oriented high level language version of the algorithm is presented.

A FORTRAN program which generates realistic data for signal analysis is described. Using the simulated data, an instruction level simulation program, and a language compiler, the algorithm is tested using an IBM 360 computer.

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I. INTRODUCTION

A. STATEMENT OF THE PROBLEM

Electronic Warfare Support Measures (ESM) is that division of Electronic Warfare (EW) concerned with exploitation of enemy use of the electromagnetic spectrum. ESM includes actions taken to search for, intercept, locate, and analyze electromagnetic radiations to support tactical military operations.

The continuously increasing density and complexity of today's electromagnetic environment, created by friendly as well as hostile forces, has fostered the requirement for more effective performance and greater sophistication in operational ESM systems. Current receiver technology has kept pace with this escalation, and signal intercept is virtually assured. However, the resulting increased data rates, existing and forecast manpower limitations, and limited reaction time available to the operational commander have imposed a requirement for faster and more accurate signal analysis, classification, and interpretation. Currently, major efforts are being put forth to achieve an effective ESM system design incorporating high speed digital computers for the purposes of automatic signal analysis and system control.

To perform effectively in present and future environments, an automated ESM system should exhibit the following characteristics:

1. Broad frequency range.
2. Simultaneous coverage of entire frequency range.
3. High probability of intercept.
4. Low false alarm rate.
5. Capability to recognize pulse or radar-like signals.
6. Capability to determine parameters associated with radar pulse trains.
7. Capability to classify signals according to type and source.
8. Direction finding capability.
9. Capability to handle data rates.
10. Capability to detect signals at ranges to radar horizon.

The successful design and development of a system exhibiting these characteristics rest upon three equally important elements: system hardware, system software, and the man-machine interface.

1. ESM Hardware/System Description

A discussion of ESM hardware as a specific problem area is beyond the scope of the thesis. A system concept compatible with the state of the art approach is presented here as a basic foundation for this study and future efforts in the ESM arena.

Conceptually, an airborne ESM system with the characteristics previously described could be implemented utilizing a combination of Instantaneous Frequency Measurement (IFM) and Superhet receivers. The IFM receivers, operating in conjunction with a high-gain pencil beam antenna, would cover the frequency bands of interest and function as the primary acquisition receivers of the system. In addition to contributing to increased system sensitivity, the high-gain pencil

beam antenna functions to reduce the instantaneous data rate and, in conjunction with azimuth centroiding, provides an accurate means of determining signal direction of arrival.

Additional data rate reduction is accomplished by routing each IFM output through a Pulse Train Separator (PTS), which functions as a signal preprocessor to separate stable pulse trains from the received data. Parameter measurements are performed on those stable trains separated and a computer word is generated to tag the associated emitter. The required capacity of the PTS in number of pulse trains separated is considered paramount to system operational effectiveness.

A computer word is generated for each pulse associated with non-stable pulse trains and those stable pulse trains exceeding the capacity of the PTS. These data are routed through a residual channel of the PTS to a high-speed digital computer for deinterleaving and parameter measurement. The parameters associated with each signal are then passed to a main frame computer for classification, threat analysis, and tactical environment update.

The IFM system employs an omni antenna for side lobe suppression to prevent signal poke-through in the receiving antenna minor lobes. The IFM sensitivity and antenna gain provide a capability for radar detection out to the radar horizon for aircraft altitudes in the neighborhood of 30,000 feet.

Priority signals, unclassifiable signals, and signals too complex for automatic analysis are automatically assigned

to digitally controlled superhet receivers for refined analysis. The superhet receivers cover the frequency bands of interest and operate in conjunction with an omni or low-gain broad beam antenna.

2. Man-Machine Interface

The human operator element of an automated ESM system, though normally an area addressed by human factors engineers, impacts heavily on the software requirements and will be discussed here in that regard.

Automation does not eliminate the requirement for human operators, but it does significantly affect the role the operator plays in the ESM system and it may seriously retard the operator's proficiency and ability to function efficiently as a back up in the event of a system failure. Thus, in deciding which system functions should be automated and which should be assigned to the human operator, it is imperative that the operator be assigned those functions that maximize the use of his capability to exercise judgment, improvise, reason inductively, and profit from experience.

The human operator must be able to control and direct the operation of the system. To do so, he must be provided relevant information by the system at the proper time, in the proper format, and on the proper displays. To obtain efficient and reliable software, these requirements must be clearly and accurately defined in the planning stage of the ESM system development.

3. ESM Software

It is felt that software represents the most troublesome problem in the design and development of an automatic ESM system. Assuming cost and risk vary directly with complexity, ESM software will account for a substantial percentage of total system costs and be the riskiest part of the system development. Since system performance hinges on software performance, implementation of an effective software test plan would serve a threefold purpose of cost reduction, risk reduction, and increased system performance.

The ability to successfully implement such a test plan requires a detailed description of the systems functions and a software modular building process. Each software module or set of modules is then integrated into the system and tested against sets or subsets of system functions according to prescribed test plans.

B. SCOPE OF THE THESIS

Simulation, via computer software, is one technique that is useful in software development and verification or testing. Realistic data, created by successful simulation of the ESM system hardware operating in a dynamic environment, utilized as inputs to a high-speed digital computer would provide the framework for implementation, testing, and optimization of ESM signal processing algorithms. The inherent ability to control the density of the electromagnetic environment through simulation would provide the capability to analyze worst case

situations, thereby enabling system designers to estimate computer capacity and reserve capacity requirements as well as data rate reduction requirements.

Given these simulation capabilities, an instruction-level simulation of the target computer provides a powerful means of debugging software and algorithm testing for accuracy and speed of operation. Applied Technology Division of Itek Corporation, Sunnyvale, California, has provided the Naval Postgraduate School with the programming support system for their ATAC computer. Instruction level simulation is only a part of this software package which includes the compiler programs for Programming Language for ATAC (PL/ATAC), a high level language developed for ATAC applications programming. Acquisition of the ATAC Programming Support System (APSS) gives impetus to the utilization of simulation as a technique for ESM software development and testing at the Naval Postgraduate School.

Total hardware simulation of the ESM system previously described is beyond the scope of the thesis. Total system simulation does, however, lend itself to a software "build" process and as a first effort of the process, three specific objectives of this work are as follows:

1. Provide a digital simulation of an IFM receiver operating in conjunction with a high-gain, pencil beam DF antenna in a dynamic radar environment.
2. Provide a digital simulation of a hard-wired pulse train deinterleaver fed by an IFM receiver.
3. Describe and demonstrate the capabilities of APSS, particularly, the PL/ATAC programming language.

The overall objective of this work is to provide a building block as a basis for follow on study of ESM system software at the Naval Postgraduate School.

II. COMPUTER SYSTEM FOR ESM SOFTWARE DEVELOPMENT

A. GENERAL DESCRIPTION

APSS was written for Applied Technology's Airborne Computer (ATAC). ATAC, specifically designed for Electronic Warfare applications, is well suited for ESM software design, development and testing. A number of features of this 16-bit, fixed point, two's complement computer permit the establishment of a meaningful baseline for determining the relative merit of algorithms applicable to computer based ESM systems. Special features of interest are:

1. 437 nanosecond major cycle time.
2. Sixteen general registers which may be used as accumulators or as index registers.
3. Microprogrammable processor, permitting optimization of the instruction set for specific applications.
4. Memory types can be mixed in any combination of low power Read Only Memory (ROM), fast bipolar Random Access Memory (RAM), and slower core memory devices.
5. Memory expandable to 65K words.
- 3 6. Four basic instruction addressing modes: register, immediate, direct, and direct indexed.
7. Three supplemental addressing modes: register indexed, immediate short, and direct register.
- 77 8. Provisions for up to 16 Direct Memory Access (DMA) channels.
- 4 9. Provisions for up to four simultaneously active DMA channels.
- 6 10. Twelve hardware priority interrupts.
- 8 11. Modular, compact construction.

Additional detailed information, including the instruction set, may be obtained from ATAC, Applied Technology Airborne Computer, Vol. I, Principles of Operation.

B. ATAC PROGRAMMING SUPPORT SYSTEM (APSS)

This software support system allows for the preparation and debugging of programs via an IBM 360/370 computer system. APSS, through simulation of the ATAC computer and associated peripheral devices, permits the applications programmer to optimize programs for execution speed and memory utilization. The APSS system is written in a combination of PL/1, FORTRAN IV, and IBM assembler language and includes the programs listed below.

- APSS Monitor
- ATAC Instruction Assembler
- ATAC Instruction-Level Simulator
- Relocating and Linking Loader
- Microcode Assembler
- Microcode-Level Simulator
- PL/ATAC Compiler
- Library Control System
- Subroutine Library

Arrangements were made with Applied Technology to use portions of APSS and consequently, the ATAC Instruction-Level Simulation Program and the PL/ATAC Compiler are currently operational on the IBM 360/67 at the Naval Postgraduate School. Information concerning the implementation of these programs and instructions pertaining to the use of the programs are contained in Appendix A. Additional APSS information is available in ATAC, Applied Technology Airborne Computer, Volume 2, ATAC Programming Support System (APSS). The documents in this volume provide a detailed description of how to set up and use the APSS system. Those documents applicable to this work are:

1. Simulation of instruction execution.
2. Tracing option of the 16 general registers. The contents of each register can be printed after the execution of an instruction.
3. Accumulated program execution time printed with each line of trace.
4. Instruction execution histograms by frequency of execution or time of execution can be created and printed.
5. Partial and total simulated memory dump options.
6. Initialization of program data arrays allowing data to be read into virtual program memory prior to program simulation.

The APSS Instruction-Level simulator program includes a large repertoire of error messages to assist the applications programmer in debugging job management and ATAC instruction usage errors.

C. PL/ATAC

PL/ATAC, Programming Language for ATAC, is a high level, block structured language designed specifically for the ATAC computer. The programming structure and numerous capabilities of the language are demonstrated in Appendix C via the Pulse Train Separator simulation program written in PL/ATAC. In addition to hardware simulation, a specific objective in writing this program was to provide programming examples of the PL/ATAC language to assist the interested programmer in learning PL/ATAC. The PL/ATAC manual (Volume 2, Section II, ATAC) is an excellent reference but may present some difficulty if used as a learning text.

PL/ATAC is a powerful language providing the programmer with programming control approaching that of assembly language

without compromising the high level of the compiler language. Additionally, ATAC assembly language may be intermixed with PL/ATAC increasing programming flexibility and control with no serious degradation of program readability.

An analysis to determine the effectiveness of applications programming in the PL/ATAC language versus ATAC assembly language was conducted by the author utilizing an assembler optimized ATAC benchmark program. The results of this analysis indicated a memory utilization increase by a factor of approximately 2.3 times the ATAC assembly language requirement and an execution time increase by approximately 2.4 times the ATAC assembly language execution time.

The PL/ATAC programmer loses some flexibility, as compared to the assembly language programmer, due to the limitation in the number of registers available (a maximum of five) to him. Also, the PL/ATAC compiler currently installed at the Naval Postgraduate School does not have a double precision capability. Double precision calculations, if required, must be accomplished in assembly language resulting in increased core space and execution time required.

III. ESM ENVIRONMENT/IFM SIMULATION

Investigation revealed the existence of a computer program [Ref. 1] designed to simulate an IFM receiver operating in conjunction with an omni antenna in a dynamic radar environment. This program was modified to incorporate a rotating DF antenna system. Routines were added to generate unstable pulse trains in the form of pulse 'jitter' and/or 'stagger'. Documentation was added to facilitate program readability.

The DATA Simulation Program generates simulated pulsed radar signal level data arriving at an IFM receiver via a rotating antenna system. Current capacity allows for simulation of up to ten emitter platforms with up to five emitters per platform. This capacity may be increased by increasing the array dimensions of the parameters associated with each emitter-platform combination. Emitter platform and IFM platform initial position information in X-Y grid coordinates, course in degrees clockwise from North, and speed in knots are set by the program input data.

Emitter characteristics, set by the program input data, consist of transmitter frequency in MHz, pulse repetition interval (PRI) in milliseconds, pulse width (PW) in microseconds, transmitter power in kilowatts, antenna gain in db, antenna beamwidth in degrees, antenna scan rate in seconds per revolution, and pulse jitter and/or stagger characteristics. A $\sin(x)/x$ antenna radiation pattern is assumed for all emitters.

IFM receiver characteristics, set by the program input data, consist of receiver threshold sensitivity in dbm, upper and lower frequency limits of the passband in MHz, minimum time measurement capability in microseconds, antenna gain in db, antenna scan rate in seconds per revolution, and antenna beamwidth in degrees.

Additional program inputs include simulation start and stop times in seconds, frequency bin size of the IFM digital processor, and random number generator calling arguments. Complete input data description is depicted in the program listing following the description of the program variables. A complete program listing is contained in Appendix B.

Utilizing the input data described, the Data Simulation Program computes the position, course, speed, and range of each emitter platform relative to the IFM platform. The closest point of approach (CPA) and distance to the CPA is computed for each emitter platform to determine if and when a platform will be within the range of detection of the IFM. Those emitters out of the detection range during the simulation are listed as such. The IFM antenna's and all emitter antennas' initial pointing directions are set by a random process utilizing a uniform random number generator. The time at which each emitter platform enters the capture area of the IFM antenna's main beam is calculated and used as a basis for computing a random transmit time for the emitters associated with that platform.

Emitter maximum effective radiated power (P_E) is then computed utilizing the mathematical model given by

$$P_E = P_t G_t \lambda^2 / 16\pi^2$$

where

P_t = Transmitted Power,

G_t = Emitter Antenna Gain,

and λ = Wavelength of Signals.

Emitter maximum range of detection (R_m) is computed by:

$$R_m = (P_E G_r / T_h)^{1/2}$$

where

G_r = IFM Antenna Gain,

and T_h = IFM Receiver Threshold Sensitivity.

For each emitter within detection range, the minimum antenna gain (S_{min}) required to break the IFM threshold sensitivity is then computed by

$$S_{min} = T_h / G_r P_r r^2$$

where

r = Distance to the Emitter.

Using the value of emitter transmit time previously calculated, the emitter antenna pointing direction is calculated. Emitter antenna gain as a function of the relative angle between the emitter antenna and the IFM antenna is then calculated and compared with the minimum antenna gain required to break the IFM threshold sensitivity. If the signal level is sufficient to break the IFM threshold, the time of arrival (TOA) of the pulse at the IFM receiver is calculated.

If the pulse does not break the IFM threshold, the emitter transmit time is incremented by the emitter PRI, the antenna position is updated, and the antenna gain calculation process is repeated until a signal level of sufficient strength is transmitted. Should no pulse be received in the scan time of one beamwidth of the IFM antenna, the subsequent scan of the receiver antenna is checked and the process continues until a pulse is received or until simulation stop time.

The time of arrival of the first pulse from each emitter is stored in an array and, upon completion of this process for all emitters, the array is searched for the lowest time of arrival. The receiver antenna bearing at this TOA is calculated and provided as output data along with the parameters of the associated emitter. The emitter platform position is updated and the time of arrival of the next pulse from this emitter is calculated and stored in the array. The array is then searched for the next lowest time of arrival and the process repeated until simulation stop time.

The output of the Data Simulation Program consists of parameters associated with each pulse received by the IFM receiver in the proper time sequence. These parameters include the frequency, pulse width, bearing and time of arrival.

This data, as prepared for the ATAC simulation programs, consists of five integers for each pulse. These integers represent the following:

1. Emitter bearing. An integer in the range of 0-359. This word represents the IFM antenna bearing at the time of arrival of the pulse.
2. Frequency bin. The bins are numbered in ascending order of frequency from 1-128. This range may be altered as desired by the programmer.
3. Pulse width. The pulse width word is an integer in the range 0-65,535. The time scaling factor is arbitrary. Current convention is that the least significant bit of the pulse width word corresponds to .1 microseconds. This allows for representation of pulse widths from 0-6.5 milliseconds to within 0.05 microseconds.
4. Time of arrival. Time of arrival is expressed as a double precision (32 bit) integer. Word 4 is the least significant half of this integer and holds time values up to 0.84 seconds with the least significant bit corresponding to 25.6 microseconds.

5. Time of arrival. The upper half of the time word holds time values up to 15.3 hours.

The Data Simulation Program produces these outputs in printed format and in a punched card format suitable for acceptance by the ATAC simulation program. A sample of the program output is contained in Appendix B following the program listing. The program flow chart is illustrated in Figures 1 and 1a.

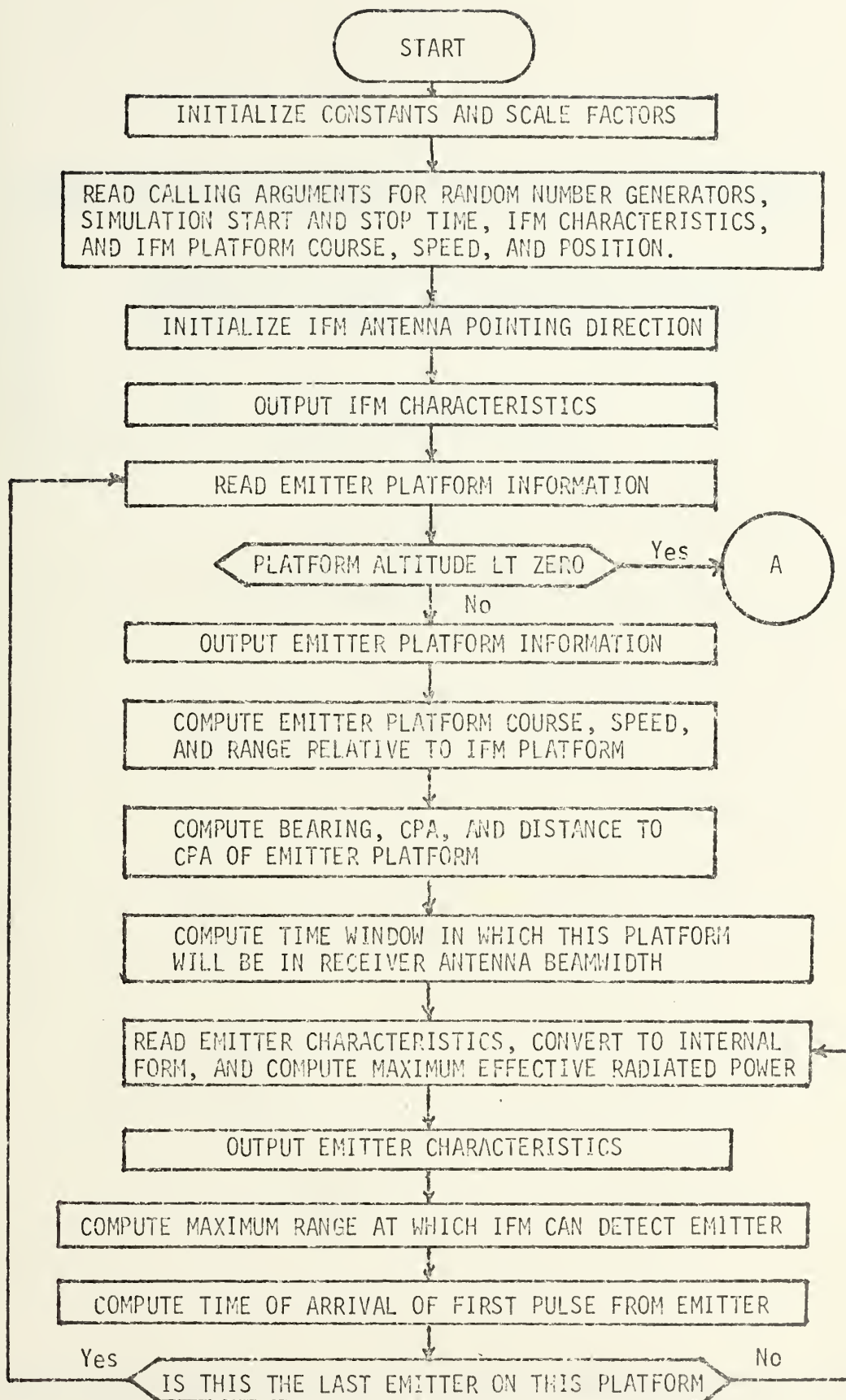


FIGURE 1. Data Simulation Program (Simplified Flowchart)

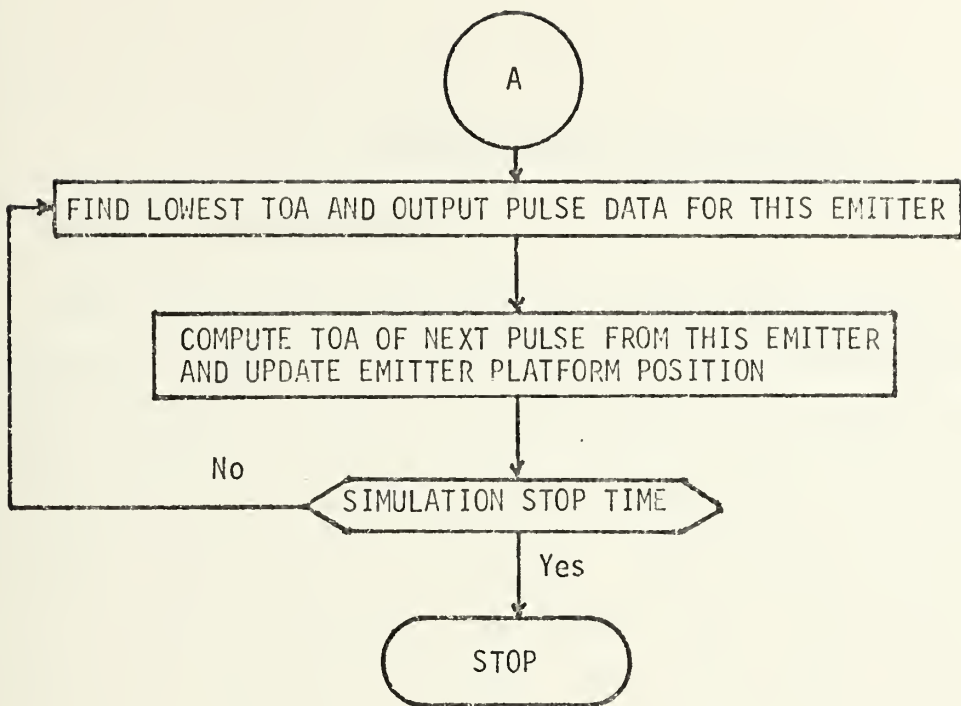


FIGURE 1a. Data Simulation Program (Simplified Flowchart)

IV. THE PULSE TRAIN SEPARATOR

A. CONCEPT OF OPERATION

Data received by the IFM are routed to the Pulse Train Separator (PTS). If a pulse train has a stable pulse repetition frequency (PRF) the PTS locks on the train and tracks the train for a period of time determined by the receiver antenna scan rate and beamwidth. The signal's pulse width (PS) and pulse repetition interval (PRI) are measured and digitized along with azimuth, time first seen and frequency. These five parameters are used to tag the associated emitter and are routed to the main frame computer for emitter classification. Direction of arrival is computed by azimuth centroiding based on the receiver antenna bearing at the time of arrival of the first and last pulse of the signal. The possibility exists that, due to $\sin(x)/x$ characteristics of emitter antennas, centroiding will be performed on ragged signal inputs. This is not considered degrading to system performance as meaningful direction of arrival information will still result.

An inherent characteristic of the Pulse Train Separator is that those stable trains with the lower values of PRI will be separated first. This feature of the PTS is also desirable as the data rate reduction role of the device is enhanced. The required capacity of the device in this role of data rate reduction has not been established and is considered a most

important factor in the overall ESM system design. Failure to provide the proper level of preprocessing could result in system software and hardware overloading and, consequently, degraded system performance.

Those stable signals exceeding the capacity of the Pulse Train Separator and non-stable signals such as frequency agile signals, jittered pulse trains, staggered pulse trains, and other exotic signals are routed through a residual signal channel of the pulse train separator to the high speed mini computer for parameter measurement. For each of these pulses received, a computer word will be generated for frequency, time of arrival, pulse width, and direction of arrival. Pulse deinterleaving and such parameter measurements as pulse repetition interval and pulse jitter and stagger characteristics will be performed by the mini computer for emitter classification.

B. PTS SIMULATION PROGRAM

The Pulse Train Separator Simulation Program is designed to simulate a hardware preprocessing system with a capability of separating up to a maximum of ten stable pulse trains from the output of an IFM receiver.

Input data to the program consist of a series of five integer words for each pulse generated by the Data Simulation Program; direction of arrival in degrees clockwise from North, frequency by bin number, double precision time of arrival (two words) in seconds with the least significant bit

corresponding to 25.6 microseconds and pulse width in seconds with the least significant bit corresponding to .1 microseconds. These data are loaded into a large input buffer prior to the start of the program simulation.

Each pulse within the buffer is compared in frequency with previously received pulses within the same receiver antenna beamwidth. If no match in frequency occurs the pulse is considered a new signal and a pointer address is stored in a New Signal Pointer File. If a frequency match occurs, the pulse width of the signals are compared. Upon obtaining a match in pulse width the matched pulse's pointer address is stored in a five element array within the New Signal Pointer File. Upon receipt of five pulses of a pulse train, the program calls a local re-entrant procedure (PRICOMP) wherein the stability of the pulse train is determined. For purposes of this simulation a pulse train is considered unstable if the variation in time interval between pulses is greater than 51.2 microseconds and/or three or more missing pulses are encountered in ragged signals. Upon determining a pulse train to be stable, the pulse repetition interval of the signal is calculated and the signal is flagged as a stable train. The number of available channels in the Pulse Train Separator is then reduced by one and the signal is zeroed in the input buffer as each additional pulse is received. Parameters associated with this pulse train are stored and the signal bearing is updated as each pulse is received through incrementation of the input buffer.

Upon receipt of the last pulse expected from the pulse train, the centroid direction of arrival is calculated and the signal information is zeroed in the New Signal Pointer File as well as in the first five input buffer locations. The time of arrival of the last pulse expected is computed utilizing the scan time for one receiver antenna bandwidth.

Non-stable signals are flagged as such and remain in the input buffer. Upon completion of incrementation through the input buffer, all stable pulses, except those from trains that remain due to PTS capacity having been exceeded, have been zeroed from the input buffer. Parameters associated with these trains are stored in sequence according to pulse repetition interval.

The pulse information remaining in the input buffer, associated with the unstable pulse trains and with pulse trains exceeding the PTS capacity, is then compressed within the input buffer, thus allowing for simulation of the mini computer signal processing algorithm. In the initial design concept of this simulation program, these remaining data, representing the Pulse Train Separator residual channel, would be processed by an ATAC assembly language program incorporated as a local procedure within the PTS program using the code selection head as described in the PL/ATAC manual.

A complete PL/ATAC program listing of the Pulse Train Separator is contained in Appendix C. An assembly language

program generated by the PL/ATAC compiler is contained in Appendix D. A program flowchart is illustrated in Figures 2 and 2a.

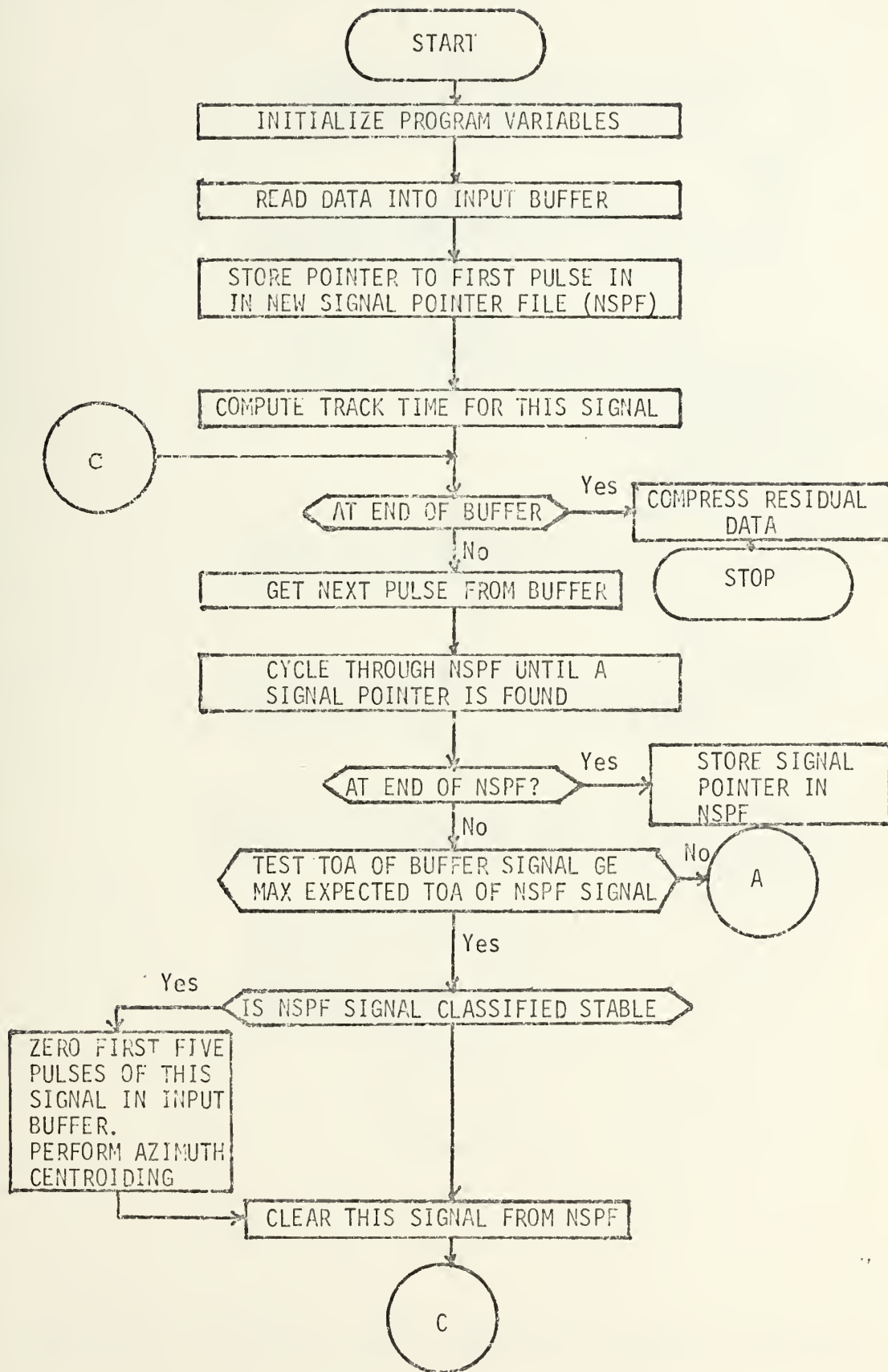


FIGURE 2. The Pulse Train Separator Program Flowchart

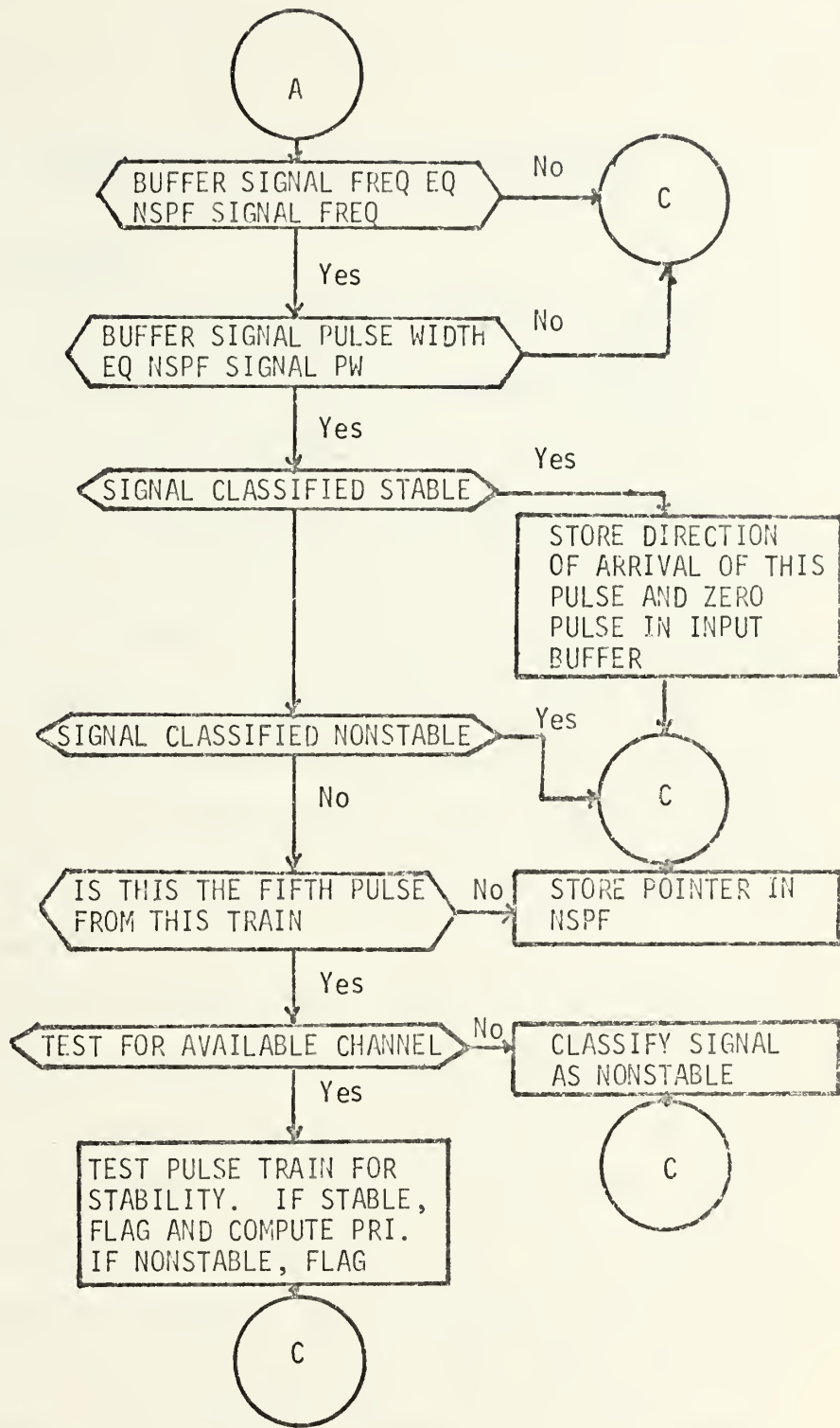


FIGURE 2a. The Pulse Train Separator Program Flowchart

V. DISCUSSION OF SIMULATION RESULTS

A. SUMMARY OF RESULTS

The Pulse Train Separator as described in Chapter IV was simulated utilizing the ATAC Programming Support System and the PL/ATAC Programming Language. The particular hardware system simulated was arbitrarily given the capability of deinterleaving and preprocessing up to nine radar pulse trains.

Input data, provided by the Data Simulation Program, consisted of a mixture of stable and non-stable pulse trains from eighteen emitters located on six platforms geographically positioned so as to be intercepted in one receiver antenna beamwidth. The receiver antenna beamwidth was arbitrarily set at five degrees.

Using a receiver antenna scan rate of two seconds per revolution, the Data Simulation Program generated a total of 365 pulses as IFM output during one antenna main beam sweep of the eighteen emitters. These pulses represent approximately 28 milliseconds of real time data. The Pulse Train Separator Simulation Program, in processing nine of the pulse trains, removed 277 pulses from the input data and routed 88 pulses through the residual channel. These 88 pulses represent the unstable pulse trains and the lower pulse repetition frequency (PRF) signals. Accurate pulse repetition interval calculations as well as accurate azimuth centroiding was performed on all preprocessed pulse trains.

B. CONCLUSIONS

The Data Simulation Program provides realistic data to the APSS programs. The program produces reasonably complex data in the form of jittered pulse trains and staggered pulse trains. Pulse trains transmitted at sufficiently low power levels are produced with missing pulses due to $\sin(x)/x$ effects.

The Pulse Train Separator Simulation Program demonstrated the use of the PL/ATAC language and functions as designed to preprocess ESM signals. This program is somewhat inefficient in its present form for two reasons. First, the limitation of five registers allocated to the programmer by the PL/ATAC compiler leads to a significant increase in program execution time as compared to a program written in ATAC assembly language. As the programmer gains experience with the PL/ATAC language, this limitation of the compiler may be overcome somewhat by programmer 'tricks' to gain use of registers.

The lack of a double precision capability in the current version of the PL/ATAC compiler installed on the IBM 360/67 at the Naval Postgraduate School created a severe handicap in writing the Pulse Train Separator Simulation Program and also contributes to program inefficiency. Each double precision calculation performed requires a program segment of assembly language and the housekeeping involved results in increased program execution time.

Even with these limitations it is felt that the simulation programs presented in this work provide the framework for ESM

software development and applications programming at the Naval Postgraduate School.

C. RECOMMENDATIONS

Based upon the experience and results obtained from this study, recommendations for follow-on study in the area of ESM software development are as follows:

1. Continue study of the Pulse Train Separator Program and optimize the basic algorithm.
2. Implement the resulting improved algorithm in ATAC assembly language.
3. Use the Data Simulation Program and optimized hardware simulation with APSS to develop and optimize ESM signal processing algorithms.
4. Continue use of the PL/ATAC compiler to develop and test hardware simulation and other programs prior to implementation in ATAC assembly language.

APPENDIX A

INSTRUCTIONS FOR INSTALLATION AND USE OF THE ATAC PROGRAMMING SUPPORT SYSTEM (APSS) ON THE IBM 360/67 OPERATING SYSTEM AT THE NAVAL POSTGRADUATE SCHOOL

A. INSTALLATION

The ATAC Programming Support System is currently loaded on the 2321 Data Cell. Data set names and their locations are as follows:

1. S1615.RDS.LOADLIB - CEL 007
2. S1616.RDS.APSS - CEL 005
3. S1615.RDS.PLATAC - CEL 001

In the event of failure of the Data Cell these data sets may be reloaded from the APSS source tape stored at the W.R. Church Computer Center under the name 'ATI.006'.

File #1 of this tape contains a description of the tape. File #2 contains examples of the JCL required to create and run APSS. The instructions contained in this file are superseded by the instructions contained here. The following JCL is required to list these files:

```
//(STANDARD JOB CARD)
//EXEC PGM=IEBGENER
//SYSPRINT DD SYSOUT=A
//SYSIN DD DUMMY
//SYSUT2 DD SYSOUT=A, DCP=(LRECL=80,RECFM=F,BLKSIZE=80)
        LABEL=(1,SL,,IN),DCB=(BLKSIZE=800,LRECL=80,
        RECFM=FB),VOL=SER=ATI006,DSN=FILE1
// EXEC PGM=IEBGENER
// SYSPRINT DD SYSOUT=A
//SYSIN DD DUMMY
//SYSUT2 DD SYSOUT=A,DCB=(LRECL=89,RECFM=F,BLKSIZE=80)
//SYSUT1 DD UNIT=(2400,,DEFER),DISP=(OLD,PASS),
        LABEL=(1,SL,,IN),DCB=(BLKSIZE=800,LRECL=80),
        RECFM=FB,VOL=SER=ATI006,DSN=FILE2
```


File #3 of the source tape contains the object module library of APSS and is required to build the APSS Instruction Level Simulator. File #4 contains the PL/ATAC Compiler object module. The following steps are required to build APSS from files 3 and 4.

1. Read files 3 and 4 from tape to disc storage. Special arrangements should be made to insure that the disc pack used has sufficient unallocated space for this step and the following step of this procedure. SPOOL3 has been made available in the past and is the disc used in the JCL of this example. The load modules from file 3 are moved into a disc file named 'S1614.RDS.LOADLIB'. File 4 is moved into a disc file named 'S1615.RDS.PLATAC'. The utility control statements may be changed to rename the files as they are being loaded onto the disc. The JCL statements required are as follows:

```
//(STANDARD JOB CARD)
//STEP1 EXEC PGM=IEHMOVE, REGION=LOOK
//SYSPRINT DD SYSOUT=A
//SYSUT1 DD UNIT=SYSDA,SPACE=(TRK,(40),,CONTIG)
//DA1 DD UNIT=2314,DSN=S1615.RDS.LOADLIB,
//SPACE=(TRK,(50,10,10),,CONTIG),
//DISP=(NEW,KEEP),VOL=SER=SPOOL3
//T1TAPE DD UNIT=(2400,,DEFER),DISP=(NEW,PASS),
//      LABEL=(3,SL,,IN),
//DCB=(DEN=2,BLKSIZE=800,LRECL=80,RECFM=FB),
//      VOL=SER=ATI006
//SYSIN DD *
COPY PDS=ATI.APSS.LOADLIB,TO=2314=SPOOL3,TODD=DA1,
//      FROMDD=T1TAPE,*
FROM=2400=(ATI006,3),RENAME=S1615.RDS.LOADLIB
COPY DSN=ATI.APSS.PLATAC,TO=2314=SPOOL3,
//      FROMDD=T1TAPE,TODD=DA2,*
FROM=2400=(ATI006,4),RENAME=S1615.RDS.PLATAC
//DA2 DD UNIT=2314,DSN=S1615.RDS.PLATAC,SPACE=(TRK,
//      (20,10),RLSE),
//DISP=(NEW,KEEP),VOL=SER=SPOOL3,LABEL=RETPD=360
/*
```


2. The Library (named 'S1615.RDS.LOADLIB') now on the disc must be built into an absolute load module of the entire APSS system. The module thus built (named 'S1615.RDS.APSS') will be stored permanently on cell 005 of the data cell. This file may also be renamed as desired. The following JCL statements are required.

```
//(STANDARD JOB CARD)
//BUILD EXEC PGM=IEWL,REGION=150K,
//PARM='OVLY,XREF,LET,LIST,SIZE=(256K,20480)'
//SYSPRINT DD SYSOUT=A
//LIBRARY DD DSN=S1615.RDS.LOADLIB,UNIT=2314,
           VOL=SER=SPOOL3,DISP=SHR
//SYSLIB DD DSNAME=3YS1.FORTLIB,DISP=SHR
//SYSLMOD DD DSNAME=S1615.RDS.APSS,UNIT=2321,
           VOL=SER=CEL005,
//DISP=(NEW,KEEP),LABEL=RETPD=360,
//SPACE=(TRK,(300,20,2),RLSE)
//SYSUT1 DD UNIT=SYSDA,SPACE=(TRK,(19,19),,CONTIG),
           SEP=SYSLMOD
//SYSLIM DD*
INCLUDE LIBRARYEPMUDLN
CHANGE MSIMEIHESAPDN
INCLUDE LIBRARY (APSSMON)
INCLUDE LIBRARY (MSIM4,ASEM5,SIM16A,SIMTR1,SIMI01,GUL)
INCLUDE LIBRARYEXPLMONN
OVERLAY A1
INSERT MSIMUL,*MSIMULA,IHENTRY,IHESAP
INSERT MINT, IN, OUT
INSERT IHEDBN, IHEXID
INSERT IHEBSM, IHECSM
INSERT IHEBSK, IHEIOX, IHEIOP, IHEDID, IHEDOB
INSERT IHEBIB, IHEDOA, IHEIOB
INSERT IHEIOA, IHEOCL
INSERT IHEBSD, IHEBSF
INSERT IHEJXS
INSERT IHEOSD, IHEOST, IHEBSI
INSERT IHEVPE, IHEDMA, IHEVFB
INSERT IHEDNC, IHEVFD, IHEVFA, IHEVPD, IHEVPB, IHEVSC
INSERT IHEVSD, IHEVFE, IHEDCN, IHEUPB
INSERT IHEVFC, IHEVPE, IHEVPG, IHEVQB, IHEVQC
INSERT IHEABN, IHEIOD, IHEIOF, IHEPRT, IHEVQA, IHESPRT
INSERT IHEBEG, IHEERR, IHESI2
INSERT MISET
OVERLAY A1
INSERT ASSEM, REWIND, REW72, DSKOUT, CARDIN, DISKIN,
           ERPRT, PRIADD
```



```

INSERT WRDATA, PRICOM, PRINOR, WRTX, REFTIT, PREF,
      ERTIT
OVERLAY A1
INSERT PARMRD, PRESIM
OVERLAY A1
INSERT SMLTR, NRMTRM, STRTSM, TPAGE
INSERT RDCRD, ABNRMT, ARTHFP, TRACE, HGRAM, HGRAMI,
      HGRAMS
INSERT IOINIT, ACTIVE, SIMTIM
INSERT DEVDTA, ACT, TIME, INT, RAND, DEADT, DEBUG
OVERLAY A2
INSERT LEVEL, DMAIOI, DMAIOA, DMAIOD, RIOIO
INSERT REMACT, DMA, DMATM, RIO, RIOTM, RIOINT, INTOLY,
      DMAINT
INSERT DTRAN, PUTACT, RANDOM
OVERLAY A1
INSERT HGPRNT
OVERLAY A1
INSERT LAODER
OVERLAY A1
INSERT LINK, ENTEXT, SLH
OVERLAY A1
INSERT PLATAC, IOPACK
OVERLAY $OBJECTEREGIONN
INSERT OBJECT, INIT, LIB, RCALPH, RCHEX, RCINT
INSERT MDATE
OVERLAY $DUMPEREGIONN
INSERT SMDUMP,PAGE
ENTRY MAIN
NAME APSS
/*

```

3. The PL/ATAC compiler object module on the disc must be moved to the data cell.

```

//(STANDARD JOB CARD)
//EXEC PGM=IEBGENER
//SYSPRINT DD SYSOUT=A
//SYSUT1 DD DISP=SHR,UNIT=2314,VOL=SER=SPOOL3,
//DSN=S1615.RDS.PLATAC
//SYSUT2 DD DISP=(NEW,KEEP),UNIT=2321,VOL=SER=CEL001,
//DCB=(RECFM=FT,BLKSIZE=7188),LABEL=RETPD=360,
//SPACE=(7188,(20,0)),DSN=S1615.RDS.PLATAC
//SYSIN DD *
GENERATE
/*

```


4. The Library ('S1615.RDS.LOADLIB') must now be moved from the disc to the data cell for permanent storage. This library contains modules necessary for successful execution of the PL/ATAC compiler. The required JCL statements to accomplish this step are as follows:

```
//(STANDARD JOB CARD)
//EXEC PGM=IEBCOPY
//SYSPRINT DD SYSOUT=A
//SYSUT1 DD DISP=SHR,UNIT=2314,VOL=SER=SPOOL3,
//DSN=S1615.RDS.LOADLIB
//SYSUT2 DD DISP=(NEW,KEEP),UNIT=2321,VOL=SER=CEL007,
//DSW=S1615.RDS.LOADLIB,SPACE=(7294,(61,0,14)),
//DCB=(RECFM=UT,BLKSIZE=7294),LABEL=RETPD=360
//SYSUT3 DD UNIT=SYSDA,SPACE=(TRK,(20,5))
//SYSUT4 DD UNIT=SYSDA,SPACE=(TRK,(20,5))
//SYSIN DD *
COPY OUTDD=SYSUT2,IWDD=SYSUT1
/*
```

B. USER INSTRUCTION

The JCL required to compile and execute PL/ATAC programs is as follows:

```
//(STANDARD JOB CARD)
//PLATAC EXEC PGM=XPLMON,PARM='ALTER',REGION=200K
//STEPLIB DD DSN=S1615.RDS.LOADLIB,UNIT=2321,DISP=SHR,
VOL=SER=CEL007
//PROGRAM DD DSN=S1615.RDS.PLATAC,UNIT=2321,VOL=SER=CEL001,
DISP=SHR
//SYSPRINT DD SYSOUT=A
//FT05F001 DD DDNAME=SYSIN
//FT06F001 DD SYSOUT=A
//FT27F001 DD DSN=&&PLCODE,SPACE=(TRK,(20,10),RLSE),
//DCB=(BLKSIZE=3520,LRECL=80,RECFM=FB),DISP=(NEW,PASS,DELETE),
//UNIT=SYSDA
//SYSIN DD *
```

⋮

PL/ATAC Program

⋮


```

//APSS EXEC PGM=APSS REGION=200K
//STEPLIB DD DSN=S1615.RDS.APSS,UNIT=2321,DISP=(OLD,KEEP) ,
//      VOL=SER=CEL005,DCB=BLKSIZE=400
//FT05F001 DD DDNAME=SYSIN
//FT06F001 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//FT08F001 DD UNIT=SYSDA,SPACE=(CYL,1),DCB=BUFNO=1
//FT09F001 DD UNIT=SYSDA,SPACE=(CYL,(7,2)),
//      DCB=(RECFM=VBS,BLKSIZE=7180,LRECL=92,BUFNO=1)
//FT10F001 DD UNIT=SYSDA,SPACE=(CYL,(7,2)),
//      DCB=(RECFM=VBS,BLKSIZE=4204,LRECL=42,BUFNO=1)
//FT20F001 DD UNIT=SYSDA,SPACE=(CYL,(7,2)),
//      DCB=(BUFNO=1,RECFM=VBS,BLKSIZE=2004,LRECL=500)
//FT27F001 DD DSN=&&PLCODE,UNIT=SYSDA,DISP=(OLD,DELETE) ,
//      DCB=BUFNO=1
//SYSPRINT DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325)
//SYSIN DD *
$END
/*

```


APPENDIX B

DATA SIMULATION PROGRAM FORTRAN LISTING AND OUTPUT

A description of program variables is included in the Data Simulation Program listing. The input data deck setup is also included in the listing.

DATA SIMULATION PROGRAM (DSP)

THIS PROGRAM SIMULATES AN IFM RECEIVER OPERATING IN CONJUNCTION WITH A HIGH-GAIN PENCIL BEAM ANTENNA IN A PULSE RADAR ENVIRONMENT. OUTPUT DATA CONSISTS OF EMITTER BEARING, FREQUENCY, TIME OF ARRIVAL (DOUBLE PRECISION), AND PULSE WIDTH OR EACH PULSE GENERATED BY THE PROGRAM.

DESCRIPTION OF VARIABLES

```

ANG(I,J)
ANGLE
ANGRF
ANGRI
ANG(I,J)
BEAM
BBIN
BNG(I)
BNGI
BW(I,J)
BWI
CCON1
CCPA
CCSEI
CCSE0
CDI
CD2
DEGR
DELE
DEFF(I,J)
DEFFMAX
DEFFMIN
DEFFREQ
DEFFSCALE
DEFGAIN
DEFGIC
DEFF(I,J)
DEFINEXT(L)
ISTGR
ANTENNA TRUE POINTING ANGLE OF THE J EMITTER ON
THE I PLATFORM
EMITTER ANTENNA ANGLE AT TRANSMIT TIME
RCVR ANTENNA BEARING AT TIME EMITTER PLATFORM
ESCAPES RCVR ANTENNA BEAMWIDTH
RCVR ANTENNA BEARING AT TIME EMITTER ENTERS BW
ANTENNA SCAN RATE OF THE J EMITTER ON THE I PLATFORM
INPUT FREQUENCY OF ANTENNA BEAM WIDTH
IFM FREQUENCY OF BIN SIZE PLATFORM
TRUE BEARING OF THE I PLATFORM
COMPUTED BEAM ATTENUATION FACTOR OF THE J EMITTER
ON THE I PLATFORM
CALCULATION VALUE OF BW(I,J)
58.9*ANTENNA DIMENSION/WAVELENGTH
CLOSEST POINT OF APPROACH OF THE I PLATFORM
INPUT VALUE OF THE COURSE OF THE IFM PLATFORM
DISTANCE TO CPA
DETECTED TO MAX RANGE AT WHICH AN EMITTER CAN BE
CONVERSION FACTOR TO CONVERT DEGREES TO RADIANS
DELAY TIME OF THE I EMITTER ON THE J PLATFORM
MAX RECEIVE FREQUENCY OF THE IFM
MINIMUM RECEIVE FREQUENCY OF THE IFM
INPUT VALUE OF EMITTER FREQU TO UNITS
CONVERSION FACTOR EMITTER ANTENNA GAINED BY IFM
INPUT VALUE FOR NUMBER OF PULSES RECEIVED BY IFM
COUNTER FOR DESIRED TYPE OF FREQUENCY VARIATION
SELECTOR FOR SUBSCRIPT OF THE SECOND OR MORE PULSES
WHEN MORE THAN ONE PULSE IS RECEIVED
SIMULTANEOUSLY BY THE IFM
CALCULATION INDEX FOR PULSE STAGGER

```

[illegible]

[illegible]

| INPUT DATA DECK STRUCTURE | | | |
|------------------------------|--------|--------------------------|------------------------------|
| INPUT CARD SEQUENCE | | | |
| IFM PLATFORM INFORMATION | | | |
| EMITTER PLATFORM INFORMATION | | | |
| EMITTER INFORMATION | | | |
| • | | | |
| • | | | |
| (UP TO FIVE EMITTERS) | | | |
| EMITTER PLATFORM INFORMATION | | | |
| EMITTER INFORMATION | | | |
| • | | | |
| • | | | |
| (UP TO TEN PLATFORMS) | | | |
| END OF DATA CARD | | | |
| INPUT CARD | FORMAT | IFM PLATFORM INFORMATION | UNITS |
| 1 | 1-10 | NUN1 | ODD INTEGER |
| | 11-20 | NGAUS | ODD INTEGER |
| | 21-30 | VL | 10 EXP(6) YDS/SEC |
| | 31-40 | TF | SECONDS |
| | 41-50 | TF | SECONDS |
| | 51-60 | TINT | MICROSECONDS |
| | 61-70 | THRESH | DBM |
| | 1-10 | XO | NAUTICAL MILES |
| | 11-20 | YO | NAUTICAL MILES |
| | 21-30 | ZO | FEET |
| | 31-40 | CSEQ | DEGREES CLOCKWISE FROM NORTH |
| | 41-50 | SPDO | KNHZ |
| | 51-60 | FMAX | MHZ |
| | 61-70 | FMIN | INTEGERS |
| | 1-10 | NBIN | DEGREES |
| | 11-20 | RSPR | SECONDS |
| | 21-30 | RGAIN | DB |
| | 31-40 | | |
| 2 | | | |
| 3 | | | |

| | | | | | | |
|-------------------|---------|--------|----------|------------------------------|--|------|
| INPUT CARD FORMAT | | | | EMITTER PLATFORM INFORMATION | | |
| CARD | COLUMNS | FORMAT | VARIABLE | UNITS | | |
| 1 | 1-10 | F10.0 | XI | NAUTICAL MILES | | 1830 |
| | 11-20 | F10.0 | YI | NAUTICAL MILES | | 1840 |
| | 21-30 | F10.0 | ZI | FEET | | 1850 |
| | 31-40 | F10.0 | CSEI | DEGREES CLOCKWISE FROM NORTH | | 1870 |
| | 41-50 | F10.0 | SPDI | DEGREES | | 1890 |
| | 51 | I1 | NMTR(I) | KNOTS | | 1900 |
| | | | | INTEGER | | 1910 |
| INPUT CARD FORMAT | | | | EMITTER INFORMATION | | 1920 |
| CARD | COLUMNS | FORMAT | VARIABLE | UNITS | | 1930 |
| 1 | 1-10 | F10.0 | FREQ | MHZ | | 1940 |
| | 11-20 | F10.0 | PRI | MILLISECONDS | | 1950 |
| | 21-30 | F10.0 | PW | MICROSECONDS | | 1960 |
| | 31-40 | F10.0 | PYTRANS | KILOWATTS | | 1970 |
| | 41-50 | F10.0 | BEAM | DEGREES | | 1980 |
| | 51-60 | F10.0 | GAIN | DB | | 1990 |
| | 61-70 | F10.0 | SCAN | SECONDS PER REVOLUTION | | 2000 |
| | 1-10 | F10.0 | PJTI | REAL NUMBER PERCENT | | 2010 |
| 2 | 11-20 | F10.6 | STGRI | REAL NUMBER PERCENT | | 2020 |
| END OF DATA | | | | NEGATIVE NUMBER IN CC 21-30 | | 2030 |
| | | | | | | 2040 |
| | | | | | | 2050 |
| | | | | | | 2060 |
| | | | | | | 2070 |
| | | | | | | 2080 |
| | | | | | | 2090 |


```

COMMON X(10),Y(10),Z(10),VX(10),VY(10),BNG(10),F(10,5),TINT,VXO,VY
10,THRESH,PI,PI2,SCNMIN,PRIN,SIGFRQ(10,5),PRI(10,5),PW(10,5),POWER(
210,5),BW(10,5),SEND(10,5),TNEXTI(10,5),SIGNAL(10,5),ANT(10,5),ANG(1
30,5),XQ,YQ,ZO,CSEQ,SPDQ,VL,TWOPI,RDEGR,TO,TF,FMAX,FMIN,BIN,PI,X
41F(10,5),PJT(10,5),STGR(10,5),IPT(10,5),ISCAN(10,5),JNEX
51(10),NUNI,NGAUS,NMTR(10),TFINI(10,5),RSPR,RRATE,RBW,RGAIN,NOUT(40
6)

```

CCCCC

INITIALIZE CONSTANTS

```

PI2 = ARSIN(1.0)
PI = 2.*PI2
TWOPI = 40.*PI2
RDEGR = 180./PI
DEGR = PI/180.
CON1 = PI*58.9*DEGR
CON2 = 10.*ALOG10(4.*PI)
TSKTS = 2025.4
YDS = 0.562611
FSCALE = 1.E6
TSCALE = 1.E3
WSCALE = 1.E-3
IJ = 0
ICT = 0

```

CCCCC

SET IFM INFORMATION

```

READ (5,17) NUNI,NGAUS,VL,TO,TF,TINT,THRESH
READ (5,18) XQ,YQ,ZO,CSEQ,SPDQ,FMAX,FMIN
READ (5,21) NBIN,RBW,RSPR,RGAIN
VL = VL*FSCALE
TINT = TINT*WSCALE
FMAX = FMAX*FSCALE
FMIN = FMIN*FSCALE
BIN = (FMAX-FMIN)/FLOAT(NBIN)

```

CCCCC

SET IFM INITIAL ANTENNA POSITION

```

CALL RANDU (NUNI,IV,RAND)
NUNI = IV
RANT = TWOPI*RAND

```



```

CCCCC
RRATE = TWOPI/RSR
2580
2590
2600
2610
2620
2630
2640
2650
2660
2670
2680
2690
2700
2710
2720
2730
2740
2750
2760
2770
2780
2790
2800
2810
2820
2830
2840
2850
2860
2870
2880
2890
2900
2910
2920
2930
2940
2950
2960
2970
2980
2990
3000
3010
3020
3030
3040
3050

OUTPUT IFM INFORMATION

WRITE (6,23) XJ,YO,ZO,CSEO,SPDO,FMIN,FMAX,BIN,NBIN
WRITE (6,24) THRESH
WRITE (6,30) RGAIN
WRITE (6,31) RSPR
CSEO = CSEO*DEGR
SPDO = SPDO*TSKTS
XO = XO*YDS
YO = YO*YDS
ZO = ZO/3.
VXO = SPDO*SIN(CSEO)
VYO = SPDO*COS(CSEO)
I = 1

GET EMITTER PLATFORM INFORMATION

1 READ (5,19) XI,YI,ZI,CSEI,SPDI,NMTR(I)

IF ALL EMITTERS HAVE BEEN ENTERED GO SORT TOA'S

IF (ZI,LT,0.) GO TO 10
WRITE (6,25) I,XI,YI,ZI,CSEI,SPDI

COMPUTE PLATFORM RELATIVE COURSE AND SPEED

CSEI = CSEI*DEGR
SPDI = SPDI*TSKTS
VXI = SPDI*SIN(CSEI)-VXO
VYI = SPDI*COS(CSEI)-VYO
XI = XI*YDS-XO
YI = YI*YDS-YO
Z(I) = SQRT(VXI**2+VYI**2)
VTI = SQRT(VXI,VYI)
RCSE = ATAN2(VXI,VYI)

COMPUTE RANGE AND BEARING TO PLATFORM
3060
3070
3080
3090
3100
3110
3120
3130
3140
3150
3160
3170
3180
3190
3200
3210
3220
3230
3240
3250
3260
3270
3280
3290
3300
3310
3320
3330
3340
3350
3360
3370
3380
3390
3400
3410
3420
3430
3440
3450
3460
3470
3480
3490
3500
3510
3520
3530
3540
3550
3560
3570
3580
3590
3600
3610
3620
3630
3640
3650
3660
3670
3680
3690
3700
3710
3720
3730
3740
3750
3760
3770
3780
3790
3800
3810
3820
3830
3840
3850
3860
3870
3880
3890
3900
3910
3920
3930
3940
3950
3960
3970
3980
3990
4000
4010
4020
4030
4040
4050

```



```

CC      R = SQR T(XI**2+YI**2)
CC      BNGI = ATAN2(XI,YI)
CC      COMPUTE PULSE TIME OF FLIGHT FOR THIS PLATFORM
CC      TOF = R/VL
CC      X(I) = XI
CC      Y(I) = YI
CC      VX(I) = V*XI
CC      VY(I) = V*YI
CC      BNG(I) = BNGI
CC      COMPUTE CPA AND DISTANCE TO CPA
CC      THETA I = ABS(PI-BNGI+RCSE)
CC      IF (THETA I.GT.TWOPI) THETA I=THETA I-TWOPI
CC      IF (THETA I.LE.PI2) GO TO 2
CC      D1 = R*COS(THETA I)
CC      CPA = R*SIN(THETA I)
CC      GO TO 3
2 CPA = R
3 JJ = NMTR(I)
CC      COMPUTE TIME WINDOW IN WHICH THIS PLATFORM WILL BE IN THE
CC      RECEIVER ANTENNA BEAMWIDTH
CC      IF (BNGI.LE.0.) BNGI=BNGI+TWOPI
CC      IF (ABS(BNGI-RANT).LE.((RBW/2.)*DEGR)) GO TO 4
CC      ANGR I = BNGI-RBW/2.*DEGR-RANT+TWOPI
CC      IF (ANGRI.LT.0.) ANGR I=ANGRI+TWOPI
CC      ANGRF = BNGI+RBW/2.*DEGR-RANT
CC      IF (ANGRF.LT.0.) ANGRF=ANGRF+TWOPI
CC      TINI = ANGR I/RRATE+TO
CC      GO TO 5
4 TINI = TO
5 TFIN = ANGRF/RRATE+TO
CC      GET EMITTER INFORMATION

```



```

CC          CC          CC          CC          CC          CC          CC          CC          CC          CC
DO 9 J=1,JJ      FREQ,PRIN,PULSE,PTRANS,BEAM,GAIN,ANT(I,J)
READ (5,18)
READ (5,20) PJTI,SIGRI
3540
3550
3560
3570
3580
3590
3600
3610
3620
3630
3640
3650
3660
3670
3680
3690
3700
3710
3720
3730
3740
3750
3760
3770
3780
3790
3800
3810
3820
3830
3840
3850
3860
3870
3880
3890
3900
3910
3920
3930
3940
3950
3960
3970
3980
3990
4000
4010

SET PARAMETERS

PJT(I,J) = PJTI
SIGRI(I,J) = SIGRI
PTRANS = PTRANS*TSSCALE
F(I,J) = FREQ*FSSCALE
PRIN = PRIN*PSCALE
PRI(I,J) = PRIN
PULSE = PULSE*WSSCALE
PW(I,J) = PULSE
TFINI(I,J) = TFIN
BW(I,J) = CON1/BEAM

COMPUTE EFFECTIVE RADIATED POWER

POWIJ = 10.*ALOG10(PTRANS)+GAIN-20.*ALOG10(FREQ)+58.3335
POWER(I,J) = POWIJ

OUTPUT PARAMETERS OF THIS EMITTER

WRITE (6,26) J,F(I,J),PRI(I,J),PW(I,J),PTRANS,GAIN,BEAM,POWER(I,J)
WRITE (6,27) ANT(I,J)

COMPUTE MAX RANGE OF DETECTION

RMAX = 10.*((POWIJ+RGAIN-THRESH)/20.)

CALCULATE THE TOA OF THE FIRST PULSE

IF (RMAX.GE.CPA) GO TO 6
TNEXT(I,J) = TFF+1.
WRITE (6,28)

```



```

4020 GO TO 9
4030 CALL RANDU (NUNI,IV,RAND)
4040 NUNI = IV
4050 ANG(I,J) = RAND*TWOPI
4060 CALL RANDU (NUNI,IV,RAND)
4070 NUNI = IV
4080
4090 IF EMITTER RANGE GREATER THAN MAX RANGE OF DETECTION GO COMPUTE
4100 DELAY TIME
4110 OTHERWISE GO COMPUTE A TRANSMIT TIME WITHIN THE EMITTER
4120 PRI OF THE TIME EMITTER WILL BE IN THE RECEIVER ANTENNA BEAM
4130
4140
4150
4160
4170
4180
4190
4200
4210
4220
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4240
4250
4260
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```

```

6 GO TO 9
  CALL RANDU (NUNI,IV,RAND)
  NUNI = IV
  ANG(I,J) = RAND*TWOPI
  CALL RANDU (NUNI,IV,RAND)
  NUNI = IV

  IF EMITTER RANGE GREATER THAN MAX RANGE OF DETECTION GO COMPUTE
  DELAY TIME
  OTHERWISE GO COMPUTE A TRANSMIT TIME WITHIN THE EMITTER
  PRI OF THE TIME EMITTER WILL BE IN THE RECEIVER ANTENNA BEAM

  IF (RMAX.LT.R) GO TO 7
  DEL = 0.
  R1 = R
  GO TO 8
7 D2 = CPA*SIN(ARCOS(CPA/RMAX))
  DEL = (D1-D2)/VTI+(R-RMAX)/VL
  R1 = RMAX
  TT = TINI+PRIN*RAND-TOF-PULSE+DEL
  IF ((TT+TOF+DEL).LT.TINI) TT=TT+PRIN

  COMPUTE RANGE ATTENUATION

  POW = POWIJ-20.*ALOG10(R1)

  COMPUTE MINIMUM EMITTER ANTENNA GAIN REQUIRED FOR RECEPTION

  SCNMN = THRESH-POW-RGAIN

  COMPUTE RECEIVED SIGNAL POWER LEVEL
  IF ANTENNA GAIN (COMPUTED IN SCAN) WAS BELOW THE REQUIRED VALUE
  FUNCTION SCAN WILL HAVE INCREMENTED TIME AT THE EMITTER PRI
  UNTIL EMITTER ANTENNA GAIN AT SUFFICIENT LEVEL OR TIME EXCEEDS
  SIMULATION STOP TIME

  SIGNAL(I,J) = POW+SCAN(I,J)

  COMPUTE TOA

```

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CC      TNEXT(I,J) = TINT*AIN((TT+TOF)/TINT)
CC      SEND(I,J) = TT
CC      SIGFRO(I,J) = F(I,J)
CC      9 CONTINUE
CC      GO GET NEXT PLATFORM INFO
CC      I = I+1
CC      GO TO I
CC      AT THIS POINT ALL OF THE EMITTER PLATFORMS AND ALL EMITTERS
CC      HAVE BEEN READ IN AND THE TOA OF THE FIRST PULSE OF EACH
CC      EMITTER HAS BEEN COMPUTED
CC      ROUTINE BELOW IS TO FIND THE TOA WITH THE LOWEST VALUE
CC
10      NTGT = I-1
11      WRITE (6,29)
11      TEMP = IF
11      L = 0
11      ITEMP = 0
CC
CC      DO 14 I=1,NTGT
CC      JJ = NMTR(I)
CC
CC      DO 13 J=1,JJ
CC      TTRY = TNEXT(I,J)
CC      IF (TEMP.LT.TTRY) GO TO 13
CC      IF (TEMP.EQ.TTRY) GO TO 12
CC      ITEMP = TTRY
CC      JTEMP = J
CC      GO TO 13
12      L = L+1
12      INEXT(L) = ITEMP
CC      JNEXT(L) = JTEMP
CC      ITEMP = I
CC      JTEMP = J
13 CONTINUE
CC

```



```

C      14 CONTINUE
C
C      IF (ITEMP.EQ.0) STOP
C
C      CONVERT PULSE DATA TO FORMAT REQUIRED BY THE ATAC COMPUTER
C      AND OUTPUT DATA ON PRINTER AND CARDS
C
15  RCVSIG = SIGNAL(ITEMP,JTEMP)
    RCVR = SIGFRQ(ITEMP,JTEMP)
    N = (RCVR-FMIN)/BIN+1.
    ICT = ICT+1
    RPW = 1.E7*PW(ITEMP,JTEMP)
    NPW = RPW
    IF ((RPW-NPW).GT.5) NPW=NPW+1
    TNEXT1 = TNEXT(ITEMP,JTEMP)
    RCVBNG = AMOD((RANT+TNEXT1*TWOPI/RSPR),TWOPI)*RDEG
    IRXB = RCVBNG
    NTIME1 = TNEXT1*1.E7/256.
    NTIME1 = MOD(NTIME,65536)
    NTIME2 = NTIME/65536
    WRITE (6,22) IRXB,ICT,N,NTIME2,NTIME1,NPW,TNEXT1
    IF (TNEXT1.GT.1F) STOP
    NOUT(IJJ+1) = IRXB
    NOUT(IJJ+2) = NTIME1
    NOUT(IJJ+3) = NTIME2
    NOUT(IJJ+4) = NTIME2
    NOUT(IJJ+5) = NPW
    IJ = IJJ+5
    IF (IJ.LT.40) GO TO 16
    IJ = 0
    WRITE(7,32)NOUT
16  IF (ICT.GE.800 ) STOP
C
C      GO CALCULATE THE TOA OF THE NEXT PULSE FROM THIS EMITTER
C
C      CALL UPDATE (ITEMP,JTEMP)
C      IF (L.EQ.0) GO TO 11
C      ITEMP = INEXT(L)
C      JTEMP = JNEXT(L)
C      L = L-1
C      GO TO 15
C

```


SUBROUTINE UPDATE (I,J)

UPDATE IS CALLED FOR CALCULATION OF ALL TOAS AFTER THE FIRST
PULSE FOR EACH EMITTER. HERE THE EMITTER IS FIRST TESTED FOR
JITTER CHARACTERISTICS. IF JITTER PRESENT COMPUTE RANDOM
PULSE INTERVAL IN JITTER BOUNDS
IF NO JITTER CHARACTERISTICS TEST FOR STAGGER. IF STAGGER
PRESENT COMPUTE PULSE INTERVAL AT PROPER STAGGER RATIO
IF NO STAGGER NORMAL CALCULATIONS ARE PERFORMED
PLATFORM RANGE AND BEARING ARE UPDATED
POWER AND TOA CALCULATIONS ARE PERFORMED

COMMON X(10), Y(10), Z(10), VX(10), VY(10), BNG(10), F(10,5), TINT, VXO, VY
10, THRESH, PI, PI2, SCNMN, PRIN, SIGFRQ(10,5), PRI(10,5), PW(10,5), POWER(1
210,5), BW(10,5), SEND(10,5), TNEXT(10,5), SIGNAL(10,5), ANT(10,5), ANG(1
310,5), XO, YO, ZO, CSEQ, SPDO, VL, TWOP, I, RDEG, DEGR, TO, IF, FMAX, FMIN, BI, TI,
41F(10,5), PJT(10,5), STGR(10,5), IPT(10,5), ISCAN(10,5), INEXT(10), JNEX
51PRIN = PRI(I,J), NGAU, NMTR(10), TFINI(10,5), RSPR, RRATE, RBW, RGA

IF (PJT(I,J).EQ.0) GO TO 1
PRICOM = PRI(I,J) - PJT(I,J) * PRI(I,J) / 2.
CALL RANDU (NUNI, IV, RAND)

NUNI = IV
PRIN = PRICOM + RAND * PJT(I,J) * PRI(I,J)

GO TO 4
1 IF (STGR(I,J).EQ.0.) GO TO 3

IF (STGR.EQ.1) GO TO 2

PRIN = PRI(I,J) - (1. - STGR(I,J)) * PRI(I,J)

GO TO 4

2 STGR = 0
PRIN = PRI(I,J) - STGR(I,J) * PRI(I,J)

GO TO 4

3 PRIN = PRI(I,J)

4 TFINI = TFINI(I,J) + PRIN

TFT = SEND(I,J) * TFINI GO TO 5

IF (TFT - RSPR - RBW * DEGR / RRATE

DTFIN = TFINI + RSPR

5 TFINI = TFT + (AINT(DT / PRIN) + 1.) * PRIN

TFINI = X(I)

XI = Y(I)

VXI = VX(I)

VYI = VY(I)

XX = XI + VXI * TT

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```

YY = YI+VYI*TT
R = SQRT(XX**2+YY**2)
POW = POWER(I,J)-20.*ALOG10(R)
SCNMIN = THRESH-POW-RGAIN
IF (SCNMIN.GE.0.) GO TO 6
SIGNAL(I,J) = POW+SCAN(I,J)
TOF = R/VL
BNG(I) = ATAN2(XX,YY)
TNEXT(I,J) = TINT+AINI*((TT+TOF)/TINT)
SEND(I,J) = TT
RETURN
6 TNEXT(I,J) = IF+1.
WRITE (6,7) I,J
RETURN
C
7 FORMAT (' EMITTER',2I3,' OUT OF RANGE.')
END

```

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```


FUNCTION SCAN (I,J)

SCAN COMPUTES THE RELATIVE ANGLE OF THE EMITTER ANTENNA AT
TRANSMIT TIME. SIN X /X ATTENUATION IS COMPUTED
IF ANTENNA GAIN IS AT OR ABOVE REQUIRED LEVEL THE PROGRAM
RETURNS TO CALLING POINT WHERE TOA WILL BE COMPUTED.
IF ANTENNA GAIN IS TO LOW TO BREAK RECEIVER THRESHOLD, TIME
IS INCREMENTED AT THE PRI RATE UNTIL SUCH TIME AS A
SIGNAL BREAKS THRESHOLD

```

COMMON X(10),Y(10),Z(10),VX(10),VY(10),BNG(10),F(10,5),TINT,VXD,VY
10,THRESH,P1,P12,SCNMIN,PRIN,SIGFRQ(10,5),PRI(10,5),PWT(10,5),POWER(
210,5),BW(10,5),SEND(10,5),TNEXT(10,5),SIGNAL(10,5),ANT(10,5),ANG(1
30,5),XQ,YO,ZQ,CSEQ,SPDO,VL,TWOPI,RDEG,DEGR,TO,TF,FMAX,FMIN,BIN,FI,
41F(10,5),PJ(10,5),STGR(10,5),IPI(10,5),ISCAN(10,5),INEXT(10),JNEX
51(10),NUN,NGAUS,NMTR(10),TFINI(10,5),RSPR,RRATE,RBW

1  ANGI = ANG(I,J)
   IFIN = TFIN(I,J)
   BWI = BW(I,J)
   BNGI = BNG(I,J)
   RATE = ANTI(I,J)
   RATE = TWOPI/RATE
   ANGLE = AMOD((ANGI+TT*RATE-PI-BNGI),TWOPI)
   IF (ANGLE.GT.PI) ANGLE=ANGLE-TWOPI
   IF (ANGLE.LT.(-PI)) ANGLE=TWOPI+ANGLE
   U = ABS(ANGLE).LE.1.E-2) GO TO 5
   SCAN = 10.*ALOG10((SIN(U)/U)**2)
   IF (SCAN.GE.SCNMIN) GO TO 4
   TT = TT+PRIN
   IF (TT.GE.TFIN) GO TO 3
   GO TO 2
3  DTFIN = TFIN-RBW*DEGR/RRATE
   TFIN = TFIN+RSPR
   TFINI(I,J) = TFIN
   TT = TT+(AINT(DT/PRIN)+1.)*PRIN
   IF (TT.GE.TF) GO TO 4
   GO TO 2
4  RETURN 0.0
5  RETURN
END

```

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OWN UNIT INFORMATION.

INITIAL POSITION: 0.0 0.0 0.0
 THRESHOLD = 70.0 DBM
 RCVR ANTENNA GAIN = 25.0 DBM
 RCVR ANTENNA SCAN = 2.00 SPR

TARGET INFORMATION

| PLATFORM | INITIAL POSITION | COURSE | SPEED | FREQUENCY COVERAGE | BINSIZE | NR | BINS |
|--------------------|--------------------|------------|--------------|------------------------|------------|------------|---------------|
| 1 | -98.0 23.0 0.0 | 0.0 | 250.0 | 8.0000E 09- 1.2000E 10 | 3.1250E 07 | | 128 |
| EMITTER | FREQ | PRI | TRANS. POWER | ANT. GAIN | BEAM WIDTH | EFF. POWER | ANT. SCAN |
| 1 | 9.5950E 09 | 4.0000E-03 | 1.5000E 03 | 2.5000E 01 | 1.0000E 00 | 5.509E 01 | CIR 1.00 SPR |
| 2 | 8.8000E 09 | 4.0000E-03 | 2.0000E 03 | 2.0000E 01 | 2.0000E 00 | 3.243E 01 | CIR 8.00 SPR |
| 3 | 9.0000E 09 | 2.0000E-03 | 5.0000E 04 | 2.5000E 01 | 5.0000E 00 | 4.9238E 01 | CIR 10.00 SPR |
| TARGET INFORMATION | | | | | | | |
| 2 | -100.0 24.0 0.0 | 0.0 | 0. | | | | |
| EMITTER | FREQ | PRI | TRANS. POWER | ANT. GAIN | BEAM WIDTH | EFF. POWER | ANT. SCAN |
| 1 | 9.5950E 09 | 4.0000E-03 | 1.5000E 03 | 2.5000E 01 | 1.0000E 01 | 5.5328E 01 | CIR 2.00 SPR |
| 2 | 8.5000E 09 | 4.0000E-03 | 1.5000E 03 | 1.5000E 01 | 1.5000E 01 | 4.5328E 01 | CIR 1.00 SPR |
| 3 | 9.5000E 09 | 4.0000E-03 | 2.5000E 03 | 1.5000E 01 | 5.0000E 00 | 4.7758E 01 | CIR 9.00 SPR |
| TARGET INFORMATION | | | | | | | |
| 3 | -100.0 25.0 0.0 | 0.0 | 15. | | | | |
| EMITTER | FREQ | PRI | TRANS. POWER | ANT. GAIN | BEAM WIDTH | EFF. POWER | ANT. SCAN |
| 1 | 8.5850E 09 | 5.0000E-04 | 1.0000E 05 | 2.0000E 01 | 7.5000E 00 | 5.1360E 01 | CIR 1.00 SPR |
| 2 | 1.0147E 10 | 7.5000E-04 | 5.0000E 02 | 3.5000E 01 | 1.5000E 00 | 5.0189E 01 | CIR 0.25 SPR |
| 3 | 8.7000E 09 | 1.2500E-03 | 5.0000E 05 | 2.0000E 01 | 2.5000E 01 | 5.6553E 01 | CIR 1.00 SPR |
| TARGET INFORMATION | | | | | | | |
| 4 | -100.0 26.0 0.0 | 0.0 | 12. | | | | |
| EMITTER | FREQ | PRI | TRANS. POWER | ANT. GAIN | BEAM WIDTH | EFF. POWER | ANT. SCAN |
| 1 | 9.1880E 09 | 5.0453E-03 | 5.0000E 05 | 1.5000E 01 | 5.0000E 00 | 6.1059E 01 | CIR 0.10 SPR |
| 2 | 9.2500E 09 | 1.0000E-03 | 5.0000E 04 | 3.0000E 01 | 5.0000E-01 | 5.6000E 01 | CIR 0.10 SPR |
| 3 | 8.6000E 09 | 1.5000E-03 | 5.0000E 04 | 2.5000E 01 | 6.0000E 00 | 5.1633E 01 | CIR 8.00 SPR |
| 4 | 0.9000E 09 | 2.0000E-03 | 3.0000E 05 | 1.0000E 01 | 5.0000E 00 | 4.417E 01 | CIR 4.00 SPR |
| 5 | 9.6000E 09 | 1.2000E-03 | 3.5000E 05 | 1.5000E 01 | 4.0000E 00 | 4.9119E 01 | CIR 2.00 SPR |
| TARGET INFORMATION | | | | | | | |
| 5 | -200.0 50.0 5000.0 | 60. | 14. | | | | |
| EMITTER | FREQ | PRI | TRANS. POWER | ANT. GAIN | BEAM WIDTH | EFF. POWER | ANT. SCAN |
| 1 | 9.0400E 09 | 5.0000E-04 | 2.5000E 05 | 1.5000E 01 | 1.0000E 01 | 4.7631E 01 | CIR 2.00 SPR |
| 2 | 9.9000E 09 | 2.0000E-03 | 3.0000E 05 | 1.5000E 01 | 1.5000E 01 | 4.8192E 01 | CIR 5.00 SPR |
| TARGET INFORMATION | | | | | | | |
| 6 | -200.0 48.0 0.0 | 0.0 | 20. | | | | |
| EMITTER | FREQ | PRI | TRANS. POWER | ANT. GAIN | BEAM WIDTH | EFF. POWER | ANT. SCAN |
| 1 | 9.2750E 09 | 1.0000E-03 | 1.5000E 04 | 2.4000E 01 | 4.0000E 00 | 5.1116E 01 | CIR 0.36 SPR |
| 2 | 9.5000E 09 | 4.0000E-03 | 5.0000E 04 | 3.0000E 01 | 5.0000E-01 | 5.5769E 01 | CIR 10.00 SPR |

DATA

| BEARING | PULSE | FREQ | BIN | DBLE | PRECISION | TIME | PW | TOA |
|---------|-------|------|-----|------|-----------|------|----|------|
| 280 | 1 | 32 | 2 | 0 | 7381 | 1 | 10 | 8999 |
| 281 | 2 | 41 | 3 | 0 | 7421 | 1 | 20 | 9064 |
| 282 | 3 | 16 | 4 | 0 | 7446 | 1 | 0 | 9080 |
| 283 | 4 | 48 | 4 | 0 | 7454 | 1 | 0 | 9105 |
| 284 | 4 | 26 | 4 | 0 | 7464 | 1 | 5 | 9125 |
| 285 | 6 | 16 | 4 | 0 | 7470 | 1 | 2 | 9154 |
| 286 | 7 | 63 | 4 | 0 | 7482 | 1 | 2 | 9166 |
| 287 | 8 | 42 | 4 | 0 | 7486 | 1 | 5 | 9186 |
| 288 | 9 | 44 | 4 | 0 | 7503 | 1 | 8 | 9215 |
| 289 | 10 | 45 | 4 | 0 | 7507 | 1 | 0 | 9222 |
| 290 | 11 | 39 | 4 | 0 | 7537 | 1 | 5 | 9232 |
| 291 | 12 | 52 | 4 | 0 | 7547 | 1 | 4 | 9250 |
| 292 | 13 | 45 | 4 | 0 | 7550 | 1 | 5 | 9335 |
| 293 | 14 | 59 | 4 | 0 | 7556 | 1 | 0 | 9351 |
| 294 | 15 | 26 | 4 | 0 | 7558 | 1 | 5 | 9359 |
| 295 | 16 | 93 | 4 | 0 | 7576 | 1 | 0 | 9405 |
| 296 | 17 | 53 | 4 | 0 | 7579 | 1 | 5 | 9417 |
| 297 | 18 | 33 | 4 | 0 | 7584 | 1 | 0 | 9427 |
| 298 | 19 | 92 | 4 | 0 | 7588 | 1 | 5 | 9436 |
| 299 | 20 | 34 | 4 | 0 | 7593 | 1 | 0 | 9445 |
| 300 | 21 | 56 | 4 | 0 | 7595 | 1 | 5 | 9446 |
| 301 | 22 | 66 | 4 | 0 | 7596 | 1 | 0 | 9453 |
| 302 | 23 | 45 | 4 | 0 | 7600 | 1 | 5 | 9468 |
| 303 | 24 | 18 | 4 | 0 | 7604 | 1 | 0 | 9483 |
| 304 | 25 | 39 | 4 | 0 | 7610 | 1 | 5 | 9494 |
| 305 | 26 | 62 | 4 | 0 | 7615 | 1 | 0 | 9504 |
| 306 | 27 | 16 | 4 | 0 | 7621 | 1 | 5 | 9511 |
| 307 | 28 | 33 | 4 | 0 | 7628 | 1 | 0 | 9514 |
| 308 | 29 | 22 | 4 | 0 | 7632 | 1 | 5 | 9513 |
| 309 | 30 | 54 | 4 | 0 | 7635 | 1 | 0 | 9534 |
| 310 | 31 | 19 | 4 | 0 | 7636 | 1 | 5 | 9544 |
| 311 | 32 | 49 | 4 | 0 | 7644 | 1 | 0 | 9554 |
| 312 | 33 | 30 | 4 | 0 | 7648 | 1 | 5 | 9559 |
| 313 | 34 | 93 | 4 | 0 | 7654 | 1 | 0 | 9590 |
| 314 | 35 | 42 | 4 | 0 | 7663 | 1 | 5 | 9606 |
| 315 | 36 | 93 | 4 | 0 | 7667 | 1 | 0 | 9616 |
| 316 | 37 | 25 | 4 | 0 | 7670 | 1 | 5 | 9634 |
| 317 | 38 | 44 | 4 | 0 | 7674 | 1 | 0 | 9669 |

APPENDIX C

PULSE TRAIN SEPARATOR SIMULATION PROGRAM PL/ATAC LISTING

Program variables are described as they are declared in the program.

SIMPLE PL/ATAC COMPILER VERSION OF FEBRUARY 1, 1974. CLOCK TIME = 9:51:33.79.

TODAY IS MARCH 18, 1975. CLOCK TIME = 22:57:29.51.

```
1 (/**** TUGGLES: $PUNCH ) 1
2 (****/) ) 2
3 ) 3
4 (/** PL/ATAC PULSE TRAIN SEPARATOR ) 4
5 ) 5
6 ) 6
7 ) 7
8 ) 8
9 ) 9
10 THIS VERSION OF PL/ATAC CONTAINS NO DOUBLE PRECISION CAPABILITY ) 10
11 NECESSITATING EMPLOYMENT OF THE 'CODE' COLLECTION HEAD WHERE ) 11
12 DOUBLE PRECISION CALCULATIONS ARE PERFORMED. ) 12
13 ) 13
14 ) 14
15 ) 15
16 PROGRAM IS DESIGNED TO SIMULATE A HARDWARE SIGNAL PREPROCESSING ) 16
17 SYSTEM WITH A CAPABILITY OF SEPARATING UP TO A MAXIMUM OF TEN ) 17
18 STABLE PULSE TRAINS FROM THE OUTPUT OF AN IFM RECEIVER. THE IFM ) 18
19 RECEIVER OPERATES IN CONJUNCTION WITH A HIGH GAIN SCANNING ) 19
20 ANTENNA. FOR SIMULATION PURPOSES, DIRECTION OF ARRIVAL (DOA) IS ) 20
21 ASSUMED TO BE OBTAINED FROM AN AZIMUTH RESOLVER GATED BY THE ) 21
22 PULSE TRAIN SEPARATOR (PTS) HARDWARE. THE OUTPUT PULSES FROM THE ) 22
23 IFM RECEIVER FED TO THE PTS ARE SIMULATED BY LOADING A SERIES ) 23
24 OF FIVE WORDS FOR EACH PULSE RECEIVED. ) 24
25 1. DOA - IN DEGREES LSB = 1 DEG ) 25
26 2. FREQUENCY - IN TERMS OF FREQUENCY BIN OF THE PULSE ) 26
27 3,4. TIME OF ARRIVAL (TOA) - DOUBLE PRECISION VALUE ) 27
28 ) 28
29 ) 29
30 5. PULSE WIDTH (PW) - LSB = .1 MICROSECOND ) 30
31 ) 31
32 UPON RECEIPT OF FIVE PULSES OF A STABLE TRAIN, THE PTS LOCKS ) 32
33 ON THE TRAIN AND CONTINUES TO TRACK THE TRAIN FOR ONE ANTENNA ) 33
34 BEAMWIDTH. THE PULSE REPETITION INTERVAL (PRI) IS COMPUTED AND ) 34
35 AZIMUTH CENTROIDING, BASED ON DOA AND TOA OF THE FIRST AND LAST ) 35
36 PULSE OF THE TRAIN, IS EMPLOYED TO COMPUTE THE BEARING OF THE ) 36
37 EMITTER ASSOCIATED WITH THE TRAIN. THE PARAMETERS ASSOCIATED ) 37
38 WITH THE TRAIN, OUTPUT TO A MAINFRAME COMPUTER FOR SIGNAL ) 38
39 CLASSIFICATION, CONSIST OF A SERIES OF SIX WORDS. ) 39
40 1. EMITTER BEARING ) 40
41 2. FREQUENCY ) 41
42 3,4. TIME FIRST SEEN ) 42
43 5. PULSE WIDTH ) 43
44 6. PULSE REPETITION INTERVAL ) 44
45 THIS OUTPUT IS SIMULATED BY STORING THE PARAMETERS IN A MEMORY ) 45
46 BUFFER. ) 46
47 RESIDUAL PULSES PASS THROUGH A SEPARATE CHANNEL OF THE PTS TO A ) 47
48 HIGH SPEED MINI COMPUTER WHERE THEY ARE PROCESSED TO DETERMINE ) 48
49 PRI, JITTER AND STAGGER CHARACTERISTICS IF THEY EXIST. THIS ) 49
50 ACTION IS SIMULATED BY COMPRESSING THE RESIDUAL DATA WITHIN THE ) 50
51 INPUT BUFFER TO BE PROCESSED BY AN ATAC ASSEMBLY LANGUAGE ) 51
52 PROGRAM WHICH MAY BE INCORPORATED USING THE CODE COLLECTION HEAD ) 52
53 ) 53
54 (/**** G L O B A L V A R I A B L E S ****/) ) 54
55 ) 55
56 (/** DESCRIPTION OF GLOBAL VARIABLES ) 56
57 ) 57
58 RAM VARIABLES ) 58
59 ) 59
60 BUF 4000 WORD INPUT BUFFER ) 60
61 NS NEW SIGNAL POINTER FILE. HOLDS POINTERS TO A MAXIMUM OF 5 ) 61
62 PULSES FOR EACH EMITTER UP TO 40 EMITTERS. ) 62
63 TMIN DOUBLE PRECISION TIME FIRST SEEN ) 63
64 NC PTS ACTIVE CHANNEL COUNTER. INITIAL VALUE DETERMINES ) 64
65 NUMBER OF CHANNELS OF PTS. MAX NO CHANNELS = 10-NC INIT ) 65
66 P1 TIME INTERVAL BETWEEN PULSES ) 66
67 PRI PULSE REPETITION INTERVAL ) 67
68 CHAN 240 WORD BUFFER IN WHICH STABLE PULSE TRAIN PARAMETERS ARE ) 68
69 STORED ) 69
70 CNDX INDEX TO CHAN ) 70
71 CINDX INDEX TO CHAN ) 71
72 DP DATA POINTER ) 72
73 SSF STABLE SIGNAL FOUND ) 73
74 NSSF NON-STABLE SIGNAL FOUND ) 74
75 ) 75
76 ) 76
77 ) 77
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```

79 (
80 ( REGISTER VARIABLES
81 (
82 ( 1,J NS FILE INDEXES
83 ( BP BUFFER POINTER
84 (
85 (
86 ( DECLARE BUF(3999)RAM
87 ( DECLARE NS(199)RAM
88 ( DECLARE TMIN(75)RAM
89 ( DECLARE (NC,PI(5),PRI(1),CNDX)RAM
90 ( DECLARE CHAN(120)RAM
91 ( DECLARE NSSF(39)RAM
92 ( DECLARE SSF(39) RAM
93 ( DECLARE CINDX(39) RAM
94 ( DECLARE (BP,1,J) REGISTER
95 ( DECLARE DP RAM
96 (
97 ( DEFINE TRUETOOL=(FALSE,TRUE)
98 (
99 (
100 ( LOCAL PROCEDURE PRICOMP
101 (
102 ( /* PRICOMP DETERMINES THE STABILITY OF A PULSE TRAIN. IF STABLE
103 ( THE PRI IS COMPUTED AND THE TRAIN IS FLAGGED AS A STABLE TRAIN.
104 ( ALL PARAMETERS EXCEPT OF ARE LOADED INTO THE CHANNEL BUFFER.
105 ( IF THE TRAIN IS NON STABLE IT IS FLAGGED AS SUCH.
106 (
107 ( DESCRIPTION OF VARIABLES
108 (
109 ( K, KK INDEXES FOR CALCULATIONS
110 ( MP MISSING PULSE COUNTER
111 ( PRITEM CALCULATION VALUE OF PRI
112 (
113 (
114 ( DECLARE (K,MP,KK,KJ,PRITEM)RAM
115 ( K=2
116 ( KK=KJ=1
117 ( MP=0
118 (
119 ( /* TEST PULSE INTERVALS TO DETERMINE STABILITY OF TRAIN. A PULSE
120 ( TRAIN IS CONSIDERED STABLE IF NO MORE THAN TWO MISSING PULSES
121 ( ARE ENCOUNTERED IN INITIAL PULSES AND VARIATIONS IN TIME
122 ( INTERVAL BETWEEN PULSES DOES NOT EXCEED +- 51.2 MICROSECONDS. */
123 (
124 ( LOOP1:
125 ( DO
126 ( TEST@MP>2
127 ( IFSO: LEAVE LOOP1
128 (
129 ( END
130 ( TEST @PI((@K-1.)/@KK IN (@PI(.@K.)/@KJ-2) TO (@PI(.@K.)/@KJ+2)
131 ( ABOVE: DO
132 ( MP=@MP+1
133 ( KK=@KK+1
134 ( REPEAT LCOPI
135 ( END
136 ( BELOW: DO
137 ( MP=@MP+1
138 ( KJ=@KJ+1
139 ( REPEAT LCOPI
140 ( END
141 ( K=@K+1
142 ( KK=@KK
143 ( KJ=1
144 ( TEST @K>4
145 ( IFNOT: REPEAT LOOP1
146 ( END
147 (
148 ( /* COMPUTE PRI AND TEST FOR ROUND OFF ERROR. IF ERROR EXISTS
149 ( ROUND UP BY 1.
150 (
151 ( /*
152 (
153 ( K=1
154 ( PRI=0
155 ( ITERATE 4 TIMES
156 ( PRI=@PRI+@PI(.@K.)
157 ( K=@K+1
158 ( END
159 ( PRITEM=@PRI
160 ( PRI=@PRI/(4+@MP)
161 ( TEST @PRITEM MOD(4+@MP)>(4+@MP)/2
162 ( IFSO: PRI=@PRI+1
163 ( END
164 (
165 ( /* LOAD PARAMETERS INTO CHANNEL BUFFER */
166 (
167 ( DP=@NS(.@1.)
168 ( CNDX=@CNDX+@
169 ( CINDX(.@1/5.)=@CNDX
170 ( CHAN(.@CNDX+1.)=@BUF(.@DP+1.)
171 ( CHAN(.@CNDX+2.)=@TMIN(.@1/5*2.)
172 ( CHAN(.@CNDX+3.)=@TMIN(.@1/5*2+1.)
173 ( CHAN(.@CNDX+4.)=@BUF(.@DP+.)
174 ( CHAN(.@CNDX+5.)=@PRI
175 (
176 ( /* FLAG PULSE TRAIN AS STABLE AND INCREMENT CHANNEL COUNTER */
177 (
178 ( SSF(.@1/5.)=TRUE
179 ( NC=@NC+1
180 ( RETURN
181 ( END LCOPI
182 (

```


| | | | |
|-----|--|-----|---------|
| 183 | (| 183 | PRICOMP |
| 184 | /" TRAIN FAILED STABILITY TEST. FLAG TRAIN AS NON STABLE "/ | 184 | PRICOMP |
| 185 | (| 185 | PRICOMP |
| 186 | NSSF(.01/5.)=TRUE | 186 | PRICOMP |
| 187 | END PRICOMP | 187 | PRICOMP |
| 188 | (| 188 | |
| 189 | (| 189 | |
| 190 | PROCEDURE MAIN | 190 | |
| 191 | (| 191 | MAIN |
| 192 | /" DESCRIPTION OF VARIABLES | 192 | MAIN |
| 193 | (| 193 | MAIN |
| 194 | BRNGF RCVR ANTENNA BEARING AT TOA OF FIRST PULSE OF A SIGNAL | 194 | MAIN |
| 195 | BRNGL RCVR ANTENNA BEARING AT TOA OF LAST PULSE OF A TRAIN | 195 | MAIN |
| 196 | CON1 DOUBLE PRECISION TIME CALCULATION VARIABLE | 196 | MAIN |
| 197 | CON2 DOUBLE PRECISION TIME CALCULATION VARIABLE | 197 | MAIN |
| 198 | II,JJ INDICES | 198 | MAIN |
| 199 | SCANTM RCVR ANTENNA SCAN TIME FOR ONE BEAMWIDTH | 199 | MAIN |
| 200 | TCAL LOWER HALF DOUBLE PRECISION TIME OF ARRIVAL | 200 | MAIN |
| 201 | TOAU UPPER HALF DOUBLE PRECISION TIME OF ARRIVAL | 201 | MAIN |
| 202 | TMAX EXPECTED TIME OF LAST PULSE INTERCEPT | 202 | MAIN |
| 203 | TMIN DOUBLE PRECISION TIME OF ARRIVAL OF FIRST PULSE | 203 | MAIN |
| 204 | TTEM DOUBLE PRECISION TIME CALCULATION VARIABLE | 204 | MAIN |
| 205 | TTEST DOUBLE PRECISION TIME CALCULATION VARIABLE | 205 | MAIN |
| 206 | /" | 206 | MAIN |
| 207 | (| 207 | MAIN |
| 208 | DECLARE(BRNGF,BRNGL(39),SCANTM(1),II,JJ,STR)AM | 208 | MAIN |
| 209 | DECLARE(TCAL,TOAU,TMAX(79),TTEST(1),TTEM(1))AM | 209 | MAIN |
| 210 | DECLARE(CON1(1),CON2(1))AM | 210 | MAIN |
| 211 | (| 211 | MAIN |
| 212 | /" INITIALIZE VARIABLES "/ | 212 | MAIN |
| 213 | (| 213 | MAIN |
| 214 | CNDX=-6 | 214 | MAIN |
| 215 | SCANTM=1120 | 215 | MAIN |
| 216 | I=0 | 216 | MAIN |
| 217 | BP=0 | 217 | MAIN |
| 218 | NC=1 | 218 | MAIN |
| 219 | (| 219 | MAIN |
| 220 | /" SET NEW SIGNAL PCINTER FILE TO NEGATIVE VALUES "/ | 220 | MAIN |
| 221 | (| 221 | MAIN |
| 222 | ITERATE 200 TIMES | 222 | MAIN |
| 223 | NS(.01.)=-1 | 223 | MAIN |
| 224 | I=@I+1 | 224 | MAIN |
| 225 | END | 225 | MAIN |
| 226 | I=0 | 226 | MAIN |
| 227 | NS(.01.)=0 | 227 | MAIN |
| 228 | (| 228 | MAIN |
| 229 | /" LOAD INPUT BUFFER FROM CARDS "/ | 229 | MAIN |
| 230 | (| 230 | MAIN |
| 231 | READ_DEC(BUF,BUF+3999) | 231 | MAIN |
| 232 | (| 232 | MAIN |
| 233 | MAINLP: | 233 | MAIN |
| 234 | % DO | 234 | MAIN |
| 235 | (| 235 | MAIN |
| 236 | /" LOAD UP TOA OF FIRST PULSE OF SIGNAL FROM INPUT BUFFER AND | 236 | MAIN |
| 237 | COMPUTE EXPECTED TIME OF ARRIVAL OF LAST PULSE BASED ON RCVR | 237 | MAIN |
| 238 | ANTENNA SCAN TIME FOR ONE ANTENNA BEAMWIDTH "/ | 238 | MAIN |
| 239 | (| 239 | MAIN |
| 240 | TCAL=@BUF(.@BP+2.) | 240 | MAIN |
| 241 | TOAU=@BUF(.@BP+3.) | 241 | MAIN |
| 242 | CCDE | 242 | MAIN |
| 243 | LDRM D,14,TCAL,2 | 243 | MAIN |
| 244 | DADD D,14,SCANTM | 244 | MAIN |
| 245 | STRM D,14,TTEM,2 | 245 | MAIN |
| 246 | END | 246 | MAIN |
| 247 | STR=@I/5*2 | 247 | MAIN |
| 248 | TMAX(.@STR.)=@TTEM | 248 | MAIN |
| 249 | TMAX(.@STR+1.)=@TTEM(.1.) | 249 | MAIN |
| 250 | TMIN(.@STR.)=@TOAU | 250 | MAIN |
| 251 | TMIN(.@STR+1.)=@TOAU | 251 | MAIN |
| 252 | COMPUTE: | 252 | MAIN |
| 253 | % DO | 253 | MAIN |
| 254 | (| 254 | MAIN |
| 255 | /" INCREMENT BUFFER POINTER AND CHECK AT BUFFER LIMIT. IF SO, GO | 255 | MAIN |
| 256 | COMPRESS RESIDUAL PULSES WITHIN INPUT BUFFER FOR FUTHER PROCESSING | 256 | MAIN |
| 257 | /" | 257 | MAIN |
| 258 | (| 258 | MAIN |
| 259 | BP=@BP+5 | 259 | MAIN |
| 260 | TEST @BP>3999 | 260 | MAIN |
| 261 | IF SO: LEAVE MAINLP | 261 | MAIN |
| 262 | END | 262 | MAIN |
| 263 | (| 263 | MAIN |
| 264 | /" IF NO SIGNAL IN BUFFER INDEXED BY BUFFER POINTER INCREMENT | 264 | MAIN |
| 265 | PCINTER AND REPEAT COMPUTE "/ | 265 | MAIN |
| 266 | (| 266 | MAIN |
| 267 | TEST @BUF(.@BP.) | 267 | MAIN |
| 268 | ZERO: REPEAT COMPUTE | 268 | MAIN |
| 269 | END | 269 | MAIN |
| 270 | I=0 | 270 | MAIN |
| 271 | J=0 | 271 | MAIN |
| 272 | SORT: | 272 | MAIN |
| 273 | % DO | 273 | MAIN |
| 274 | (| 274 | MAIN |
| 275 | /" IF AT END OF NEW SIGNAL POINTER FILE, LEAVE SORT "/ | 275 | MAIN |
| 276 | (| 276 | MAIN |
| 277 | TEST @I>195 | 277 | MAIN |
| 278 | IF SO: LEAVE SORT | 278 | MAIN |
| 279 | END | 279 | MAIN |
| 280 | (| 280 | MAIN |
| 281 | /" IF NEW SIGNAL POINTER FILE INDEXED BY I IS EMPTY INDEX AND | 281 | MAIN |
| 282 | CHECK NEXT LOCATION "/ | 282 | MAIN |
| 283 | (| 283 | MAIN |
| 284 | TEST @NS(.01.) | 284 | MAIN |
| 285 | NEG: DO | 285 | MAIN |
| 286 | I=@I+5 | 286 | MAIN |
| 287 | REPEAT SORT | 287 | MAIN |


```

288 ( END
289 ( END
290 (
291 (/" COMPARE TOA OF BUFFER SIGNAL WITH TMAX OF SIGNAL FOUND IN NEW
292 ( SIGNAL POINTER FILE. IF TOA OF BUFFER SIGNAL IS GREATER THAN
293 ( OR EQUAL TO TMAX, TEST FOR STABLE SIGNAL IN NS POINTER FILE.
294 ( IF SIGNAL HAS BEEN CLASSIFIED STABLE, RESET ACTIVE CHANNEL
295 ( COUNTER AND SSF FLAG, ZERO FIRST FIVE PULSES FROM THIS SIGNAL
296 ( IN INPUT BUFFER AND COMPUTE EMITTER BEARING USING FIRST AND
297 ( DOA OF SIGNAL "/
298 (
299 ( CON1=@BUF(.(@BP+2).)
300 ( CLN1(.1.)=@BUF(.(@BP+3).)
301 ( CON2=@TMAX(.(@I/5*2).)
302 ( CON2(.1.)=@TMAX(.(@I/5*2+1).)
303 ( CODE
304 ( LDRM D,14,CON1,2
305 ( LDRM D,7,CON2,2
306 ( DSUB R,7,14
307 ( STRM D,7,TTEST,2
308 (
309 ( END
310 ( TEST @TTEST
311 ( NEG @TTEST: DO
312 ( TEST @SSF(.(@I/5).)
313 ( POS: DO
314 ( NC=@NC-1
315 ( SSF(.(@I/5).)=FALSE
316 ( BRNGF=@BUF(.(@NS(.@I.)).)
317 ( II=0
318 ( ITERATE 5 TIMES
319 ( JJ=0
320 ( ITERATE 5 TIMES
321 ( EUF(.(@NS(.@I+@II).)+@JJ.)=0
322 ( JJ=@JJ+1
323 ( END
324 ( II=@II+1
325 ( END
326 ( CHAN(.@CINDX(.@I/5.))=(@BRNGF+@BRNGL(.@I/5.))/2
327 ( END
328 (
329 ( END
330 (/" IF NS POINTER FILE SIGNAL HAS BEEN CLASSIFIED AS NON STABLE
331 ( RESET NSSF FLAG "/
332 (
333 ( NSSF(.(@I/5.)=FALSE
334 (
335 (/" CLEAR THE SIGNAL FROM NS POINTER FILE "/
336 (
337 ( II=0
338 ( ITERATE 5 TIMES
339 ( NS(.(@I+@II.))=-1
340 ( II=@II+1
341 ( END
342 ( I=@I+5
343 ( REPEAT SORT
344 ( END
345 ( END
346 ( TEST @BUF(.(@BP+1).)=@BUF(.(@NS(.@I.))+1).)
347 ( IFNOT: DO
348 ( I=@I+5
349 ( REPEAT SCKT
350 ( END
351 ( END
352 (/" TEST FOR MATCH BETWEEN BUFFER SIGNAL FREQ AND PW AND NS SIGNAL
353 ( POINTER FILE FREQ AND PW. IF BOTH DO NOT MATCH REPEAT SORT.
354 ( IF MATCH OCCURS AND SIGNAL HAS BEEN CLASSIFIED AS STABLE,
355 ( ZERO BUFFER SIGNAL AND REPEAT COMPUTE. IF SIGNAL HAS BEEN
356 ( CLASSIFIED AS NON STABLE, REPEAT COMPUTE. "/
357 (
358 ( TEST @BUF(.(@BP+4).)=@BUF(.(@NS(.@I.))+4).)
359 ( IFNOT: DO
360 ( I=@I+5
361 ( REPEAT SORT
362 ( END
363 ( END
364 ( TEST @SSF(.(@I/5).)
365 ( NONZERO: DO
366 ( BRNGL(.@I/5.)=@BUF(.@BP.)
367 ( II=0
368 ( ITERATE 5 TIMES
369 ( BUF(.(@BP+@II.))=0
370 ( II=@II+1
371 ( END
372 ( REPEAT COMPUTE
373 ( END
374 ( END
375 ( TEST @NSSF(.(@I/5.))=TRUE
376 ( IFSO: REPEAT COMPUTE
377 ( END
378 (
379 (/" IF SIGNAL HAS NOT BEEN CLASSIFIED, STORE BUFFER POINTER IN
380 ( NS POINTER FILE. IF THIS IS THE FIFTH PULSE ASSOCIATED WITH
381 ( THIS TRAIN, COMPUTE TIME INTERVALS BETWEEN PULSES AND CALL
382 ( FOR PK1 COMPUTATION. "/
383 (
384 ( ITERATE 3 TIMES
385 ( J=@J+1
386 ( TEST @NS(.(@I+@J.))
387 ( NEG: DO
388 ( NS(.(@I+@J.))=@BP
389 ( REPEAT COMPUTE
390 ( END
391 ( END
392 ( END

```


| | | |
|-------|--|------------|
| 393 (| NS(.@I+4.)=@BP |) 393 MAIN |
| 394 (| TEST @NC==10 |) 394 MAIN |
| 395 (| IFSO: DD |) 395 MAIN |
| 396 (| NSSF(.@I/5.)=TRUE |) 396 MAIN |
| 397 (| REPEAT COMPUTE |) 397 MAIN |
| 398 (| END |) 398 MAIN |
| 399 (| END |) 399 MAIN |
| 400 (| J=1 |) 400 MAIN |
| 401 (| ITERATE 4 TIMES |) 401 MAIN |
| 402 (| PI(.@J.)=@BUF(.(@NS(.@I+@J.)+2.)-@BUF(.(@NS(.@I+@J-1.)+2.) |) 402 MAIN |
| 403 (| TEST @PI(.@J.) |) 403 MAIN |
| 404 (| NEG: PI(.@J.)=@PI(.@J.)+32767 |) 404 MAIN |
| 405 (| END |) 405 MAIN |
| 406 (| J=@J+1 |) 406 MAIN |
| 407 (| END |) 407 MAIN |
| 408 (| CALL PRICOMP() |) 408 MAIN |
| 409 (| REPEAT COMPUTE |) 409 MAIN |
| 410 (| END SCRT |) 410 MAIN |
| 411 (| I=0 |) 411 MAIN |
| 412 (| CYCLE |) 412 MAIN |
| 413 (| TEST @NS(.@I.) |) 413 MAIN |
| 414 (| NEG: DD |) 414 MAIN |
| 415 (| NS(.@I.)=@BP |) 415 MAIN |
| 416 (| REPEAT MAINLP |) 416 MAIN |
| 417 (| END |) 417 MAIN |
| 418 (| END |) 418 MAIN |
| 419 (| TEST(I=@I+5)>195 |) 419 MAIN |
| 420 (| IFSO: DD |) 420 MAIN |
| 421 (| DUMP_DEC(NS,NS+195) |) 421 MAIN |
| 422 (| RETURN |) 422 MAIN |
| 423 (| END |) 423 MAIN |
| 424 (| END |) 424 MAIN |
| 425 (| END |) 425 MAIN |
| 426 (| END COMPUTE |) 426 MAIN |
| 427 (| END MAINLP |) 427 MAIN |
| 428 (| |) 428 MAIN |
| 429 (| /" ROUTINE BELCW COMPRESSES RESIDUAL PULSES WITHIN THE INPUT |) 429 MAIN |
| 430 (| BUFFER. AN ATAC ASSEMBLY LANGUAGE PROGRAM TO PROCESS THIS |) 430 MAIN |
| 431 (| RESIDUAL MAY BE INCORPORATED BETWEEN THE END AND END MAIN |) 431 MAIN |
| 432 (| STATEMENTS UTILIZING THE CODE COLLECTION HEAD. "/ |) 432 MAIN |
| 433 (| |) 433 MAIN |
| 434 (| BP=0 |) 434 MAIN |
| 435 (| ZEROCK: |) 435 MAIN |
| 436 (| CYCLE |) 436 MAIN |
| 437 (| TEST @BUF(.@BP+1.) |) 437 MAIN |
| 438 (| ZERD: DD |) 438 MAIN |
| 439 (| DP=@BP |) 439 MAIN |
| 440 (| LEAVE ZEROCK |) 440 MAIN |
| 441 (| END |) 441 MAIN |
| 442 (| END |) 442 MAIN |
| 443 (| BP=@BP+5 |) 443 MAIN |
| 444 (| END ZEROCK |) 444 MAIN |
| 445 (| CYCLE |) 445 MAIN |
| 446 (| BP=@BP+5 |) 446 MAIN |
| 447 (| TEST @BP>3995 |) 447 MAIN |
| 448 (| IFSO: DD |) 448 MAIN |
| 449 (| RETURN |) 449 MAIN |
| 450 (| END |) 450 MAIN |
| 451 (| END |) 451 MAIN |
| 452 (| TEST @BUF(.(@BP+1.)==0 |) 452 MAIN |
| 453 (| IFNOT: DD |) 453 MAIN |
| 454 (| I=0 |) 454 MAIN |
| 455 (| ITERATE 5 TIMES |) 455 MAIN |
| 456 (| BUF(.(@DP+@I.)=@BUF(.(@BP+@I.) |) 456 MAIN |
| 457 (| BUF(.(@BP+@I.)=0 |) 457 MAIN |
| 458 (| I=@I+1 |) 458 MAIN |
| 459 (| END |) 459 MAIN |
| 460 (| DP=@DP+5 |) 460 MAIN |
| 461 (| END |) 461 MAIN |
| 462 (| END |) 462 MAIN |
| 463 (| END |) 463 MAIN |
| 464 (| END MAIN |) 464 MAIN |
| 465 (| |) 465 MAIN |
| 466 (| |) 466 MAIN |
| 467 (| /"**** LINK MODULE FOR EXECUTING PL/ATAC PROGRAM. |) 467 MAIN |
| 468 (| \$EXECUTIVE |) 468 MAIN |
| 469 (| ****"/ |) 469 MAIN |
| 470 (| MODULE LINK |) 470 MAIN |
| 471 (| DECLARE PROCSTK(64)RAM |) 471 LINK |
| 472 (| RD = PROCSTK |) 472 LINK |
| 473 (| CALL MAIN() |) 473 LINK |
| 474 (| END LINK |) 474 LINK |
| 475 (| EOF EOF EOF EOF EOF EOF EOF EOF |) 475 LINK |

APPENDIX D

PULSE TRAIN SEPARATOR SIMULATION PROGRAM ASSEMBLY LANGUAGE LISTING AND OUTPUT

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|--|---------|
| | | RAM VARIABLES | |
| | | 4000 WORD INPUT BUFFER | 55 |
| | | NEW SIGNAL POINTER FILE. HOLDS POINTERS TO A MAXIMUM OF 5 | 56 |
| | | PULSES FOR EACH EMITTER UP TO 40 EMITTERS. | 57 |
| | | DOUBLE PRECISION TIME FIRST SEEN | 58 |
| | | PTS ACTIVE CHANNEL COUNTER. INITIAL VALUE DETERMINES | 59 |
| | | NUMBER OF CHANNELS OF PTS. MAX NO CHANNELS = 10-NC INIT | 60 |
| | | TIME INTERVAL BETWEEN PULSES | 61 |
| | | PULSE REPTITION INTERVAL | 62 |
| | | 240 WORD BUFFER IN WHICH STABLE PULSE TRAIN PARAMETERS ARE | 63 |
| | | STORED | 64 |
| | | CNDX INDEX TO CHAN | 65 |
| | | CINDX INDEX TO CHAN | 66 |
| | | DATA POINTER | 67 |
| | | STABLE SIGNAL FOUND | 68 |
| | | NSSF NON-STABLE SIGNAL FOUND | 69 |
| | | REGISTER VARIABLES | 70 |
| | | I,J NS FILE INDEXES | 71 |
| | | BP BUFFER POINTER | 72 |
| | | "/ | 73 |
| | | DECLARE BUF(3999)RAM | 74 |
| | | ORG RAM | 75 |
| | | DS 4000 | 76 |
| | | DECLARE NS(199)RAM | 77 |
| | | DS 200 | 78 |
| | | DECLARE TMIN(79)RAM | 79 |
| | | DS 80 | 80 |
| | | DECLARE INC,PI(51,PRI(1),CNDX)RAM | 81 |
| | | NC DS 1 | 82 |
| | | PI DS 5 | 83 |
| | | PRI DS 2 | 84 |
| | | CNDX DS 1 | 85 |
| | | DECLARE CHAN(120)RAM | 86 |
| | | CHAN DS 121 | 87 |
| | | DECLARE NSSF(39)RAM | 88 |
| | | NSSF DS 40 | 89 |
| | | DECLARE SSF(39) RAM | 90 |
| | | SSF DS 40 | 91 |
| | | DECLARE CINDX(39) RAM | 92 |
| | | CINDX DS 40 | 93 |
| | | DECLARE (BP,I,J)REGISTER | 94 |
| | | DP RAM | 95 |
| | | DP DS 1 | 96 |
| 2000 | | | 100 |
| 2FA0 | | | 101 |
| 3068 | | | 102 |
| 3088 | | | 103 |
| 3089 | | | 104 |
| 30BF | | | 105 |
| 30C1 | | | 106 |
| 30C2 | | | 107 |
| 3138 | | | |
| 3163 | | | |
| 318B | | | |
| 3185 | | | |

| LJC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|-----------------|---------|
| 000E | C106 0012 | BRC I,6,a5 | 161 |
| 0010 | C107 00F1 | BRC I,7,a1 | 162 |
| | | END | 163 |
| | | TEST | 164 |
| 0012 | E202 31B4 | EQU \$ | 165 |
| 0014 | 6FF2 | EQU \$ | 166 |
| 0015 | E324 30B9 | LDR D,2,0+K | 167 |
| 0017 | 08F3 | ADD IS,2,-1 | 168 |
| 0018 | 08F3 31B6 | LDR DX,4,0+PI,2 | 169 |
| 001A | E205 | SHD RA,3,16 | 170 |
| 001B | FC53 | LDR O,5,0+KK | 171 |
| 001D | E205 31B4 | DIV R,3,5 | 172 |
| 001F | E357 30B9 | LDR D,5,0+K | 173 |
| 0020 | D8F6 | LDR DX,7,0+PI,5 | 174 |
| 0022 | E208 31B7 | SHD RA,6,16 | 175 |
| 0023 | 6FE7 | LDR D,8,0+KJ | 176 |
| 0024 | E208 31B4 | DIV R,6,8 | 177 |
| 0026 | E383 30B9 | ADD IS,7,-2 | 178 |
| 0028 | D8F2 | LDR D,8,0+K | 179 |
| 0029 | E205 31B7 | LDR DX,3,0+PI,8 | 180 |
| 002B | FC52 | SHD RA,2,16 | 181 |
| 002C | 6023 | LDR D,5,0+KJ | 182 |
| 002D | E035 | DIV R,2,5 | 183 |
| 002E | E036 | ADD IS,3,2 | 184 |
| 002F | 5476 | LDR R,5,3 | 185 |
| 0030 | 8854 | LDR R,5,5 | 186 |
| | | SUB R,6,7 | 187 |
| | | CBL R,4,5 | 188 |
| | | ABOVE: 00 | 189 |
| 0031 | C106 0041 | EQU \$ | 190 |
| 0033 | E205 31B5 | BRC I,6,a9 | 191 |
| 0035 | 6015 | MP=AMP+1 | 192 |
| 0036 | 9C05 31B5 | D,5,0+MP | 193 |
| | | ADD IS,5,1 | 194 |
| 0038 | E205 31B6 | LDR D,5,0+MP | 195 |
| 003A | 6015 | KK=KK+1 | 196 |
| 003B | 9C05 31B6 | D,5,0+KK | 197 |
| | | ADD IS,5,1 | 198 |
| 003D | C107 0008 | LDR D,5,0+KK | 199 |
| | | REPEAT LOOP1 | 200 |
| | | I,7,a0 | 201 |
| | | BRC END | 202 |
| 003F | C107 004F | DO I,7,a6 | 203 |
| 0041 | C105 004F | BRC \$ | 204 |
| 0043 | E205 31B5 | EQU I,3,a10 | 205 |
| 0045 | 6015 | BRC MP=AMP+1 | 206 |
| 0046 | 9C05 31B5 | LDR D,5,0+MP | 207 |
| | | ADD IS,5,1 | 208 |
| 0048 | E205 31B7 | LDR D,5,0+MP | 209 |
| 004A | 6015 | STR KJ=AKJ+1 | 210 |
| | | ADD O,5,0+KJ | 211 |
| | | I,5,5,1 | 212 |
| | | ADD | 213 |

CARDNUM

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|---|---------|
| 004B | 9C05 31B7 | STR D,5,0+KJ | 214 |
| 004D | C1C7 000B | REPEAT LOOP1 | 215 |
| | | I,7,20 | 216 |
| | | END | 217 |
| | | END | 218 |
| | | EQU \$ | 219 |
| | | EQU \$ | 220 |
| 004F | E205 31B4 | K=@K+1 | 221 |
| 0051 | 6015 | LDR D,5,0+K | 222 |
| 0052 | 9C05 31B4 | ADD IS,5,1 | 223 |
| | | STR D,5,0+K | 224 |
| 0054 | E205 31B7 | KK=@KJ | 225 |
| 0056 | 9C05 31B6 | LDR D,5,0+KJ | 226 |
| | | STR D,5,0+KK | 227 |
| 0058 | 4015 | LDR IS,5,1 | 228 |
| 0059 | 9C05 31B7 | STR D,5,0+KJ | 229 |
| 005B | E205 31B4 | LDR D,5,0+K | 230 |
| 005D | 2045 | IS,5,4 | 231 |
| | | CMP REPEAT LOOP1 | 232 |
| | | IFNOT | 233 |
| | | EQU \$ | 234 |
| | | BRC I,1,214 | 235 |
| | | BRC I,7,20 | 236 |
| | | END | 237 |
| 0062 | C101 0062 | EQU \$ | 238 |
| 0060 | C107 000B | EQU \$ | 239 |
| | | END | 240 |
| | | COMPUTE PRI AND TEST FOR ROUND OFF ERROR. IF ERROR EXISTS | 241 |
| | | ROUND UP BY 1. | 242 |
| | | /" | 243 |
| | | "/ | 244 |
| | | ." | 245 |
| | | ." | 246 |
| | | ." | 247 |
| | | ." | 248 |
| | | ." | 249 |
| 0062 | 4015 | K=1 | 250 |
| 0063 | 9C05 31B4 | LDR IS,5,1 | 251 |
| | | STR D,5,0+K | 252 |
| 0065 | 9C0C 30BF | PRI=0 | 253 |
| | | STR D,12,0+PRI | 254 |
| | | ITERATE 4 TIMES | 255 |
| | | PRI=@PRI+@PI(.2K.) | 256 |
| 0067 | 4FB5 | LDR IS,5,-5 | 257 |
| 0068 | C107 0077 | BRC I,7,216 | 258 |
| | | EQU \$ | 259 |
| 006A | E206 31B4 | D,6,0+K | 260 |
| 006C | E207 30BF | LDR D,7,0+PRI | 261 |
| 006E | 8367 30B9 | ADD DX,7,0+PRI,6 | 262 |
| 0070 | 9C07 30BF | STR D,7,0+PRI | 263 |
| | | K=@K+1 | 264 |
| 0072 | E207 31B4 | LDR D,7,0+K | 265 |
| 0074 | 6017 | ADD IS,7,1 | 266 |
| 0075 | 9C07 31B4 | STR D,7,0+K | 267 |
| | | END | 268 |
| 0077 | C905 006A | EQU \$ | 269 |
| | | IBN I,5,215 | 270 |

| LUC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|-----------------------------------|---------|
| 0079 | E207 30BF | PRITEM=@PRI | 267 |
| 0078 | 9C07 31B8 | LDR D,7,0+PRI | 268 |
| | | STR D,7,0+PRITEM | 269 |
| | | FFI=@PRI/(4+@MP) | 270 |
| 007D | E207 31B5 | LDR D,7,0+MP | 271 |
| 007F | 6047 30BF | ADD IS,7,4 | 272 |
| 0080 | E203 30BF | LDR D,3,0+PRI | 273 |
| 0082 | D8F2 | SHD RA,2,16 | 274 |
| 0083 | FC72 | DIV R,2,7 | 275 |
| 0084 | 9C03 30BF | STR D,3,0+PRI | 276 |
| | | TEST @PRITEM MOD(4+@MP)>(4+@MP)/2 | 277 |
| 0086 | E202 31B5 | LDR D,2,0+MP | 278 |
| 0088 | 6042 | ADD IS,2,4 | 279 |
| 0089 | E204 31B8 | LDR D,4,0+PRITEM | 280 |
| 008B | D8F3 | SHD RA,3,16 | 281 |
| 008C | FC23 | DIV R,3,2 | 282 |
| 008D | E204 31B5 | LDR D,4,0+MP | 283 |
| 008F | 6044 | ADD IS,4,4 | 284 |
| 0090 | E045 | LDR R,5,4 | 285 |
| 0091 | D8F4 | SHD RA,4,16 | 286 |
| 0092 | 4026 | LDR IS,6,2 | 287 |
| 0093 | FC04 | DIV R,4,6 | 288 |
| 0094 | B053 | CMP R,3,5 | 289 |
| | | IFSO: PRI=@PRI+1 | 290 |
| 0095 | C106 009C | EQU I,6,@20 | 291 |
| 0097 | E206 30BF | BRC D,6,0+PRI | 292 |
| 0099 | 6016 | LDR IS,6,1 | 293 |
| 009A | 9C06 30BF | ADD STR D,6,0+PRI | 294 |
| | | END EQU \$ | 295 |
| 009C | E367 2FA0 | EQU \$ | 296 |
| 009D | 9C07 31B3 | DP=@NSI(.@I.) | 297 |
| | | LDR R,6,10 | 298 |
| | | LDR DX,7,0+NS,6 | 299 |
| | | STR D,7,0+DP | 300 |
| 00A1 | E207 30C1 | CNDX=@CNDX+6 | 301 |
| 00A3 | 6067 | LDR D,7,0+CNDX | 302 |
| 00A4 | 9C07 30C1 | STR D,7,0+CNDX | 303 |
| | | CINDX(.@I/5.)=@CNDX | 304 |
| 00A6 | E0A8 | LDR R,8,10 | 305 |
| 00A7 | D8F7 | SHD RA,7,16 | 306 |
| 00A8 | 4052 | LDR IS,2,5 | 307 |
| 00A9 | FC27 | DIV R,7,2 | 308 |
| 00AA | E202 30C1 | LDR D,2,0+CNDX | 309 |
| 00AC | 9B82 31B8 | STR DX,2,0+CINDX,8 | 310 |
| | | CHAN(.(@CNDX+1).)=@BUF(.(@DP+1).) | 311 |
| 00AE | E202 30C1 | LDR D,2,0+CNDX | 312 |
| 00B0 | 6012 | ADD IS,2,1 | 313 |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|---|---------|
| 00B1 | E203 31B3 | LDR D,3,0+DP | 320 |
| 00B3 | 6013 | IS,3,1 | 321 |
| 00B4 | E334 2000 | DX,4,0+BUF,3 | 322 |
| 00B6 | 9B24 30C2 | LDR DX,4,0+CHAN,2 | 323 |
| | | CHAN(,@CNDX+2,)=@IMIN(,@I/5*2,). | 324 |
| 00B8 | E204 30C1 | LDR D,4,0+CNDX | 325 |
| 00BA | 6024 | IS,4,2 | 326 |
| 00BB | E0A6 | R,6,10 | 327 |
| 00BC | D8F5 | RA,5,16 | 328 |
| 00BD | 4057 | IS,7,5 | 329 |
| 00BE | FC75 | R,5,7 | 330 |
| 00BF | CD06 | I,6,2 | 331 |
| 00C1 | E367 30B8 | DX,7,0+TMIN,6 | 332 |
| 00C3 | 9B47 30C2 | STR DX,7,0+CHAN,4 | 333 |
| | | CHAN(,@CNDX+3,)=@IMIN(,@I/5*2+1,). | 334 |
| 00C5 | E207 30C1 | LDR D,7,0+CNDX | 335 |
| 00C7 | 6037 | IS,7,3 | 336 |
| 00C8 | E0A3 | R,3,10 | 337 |
| 00C9 | D8F2 | RA,2,16 | 338 |
| 00CA | 4054 | IS,4,5 | 339 |
| 00CB | FC42 | R,2,4 | 340 |
| 00CC | CD03 | I,3,2 | 341 |
| 00CE | 6013 | IS,3,1 | 342 |
| 00CF | E334 30B8 | DX,4,0+TMIN,3 | 343 |
| 00D1 | 9B74 30C2 | STR DX,4,0+CHAN,7 | 344 |
| | | CHAN(,@CNDX+4,)=@BUF(,@DP+4,). | 345 |
| 00L3 | E204 30C1 | LDR D,4,0+CNDX | 346 |
| 00L5 | 6044 | IS,4,4 | 347 |
| 00L6 | E205 31B3 | ADD D,5,0+DP | 348 |
| 00D8 | 6045 | IS,5,4 | 349 |
| 00D9 | E356 2000 | DX,6,0+BUF,5 | 350 |
| 00LB | 9B46 30C2 | LDR DX,6,0+CHAN,4 | 351 |
| | | CHAN(,@CNDX+5,)=@PRI | 352 |
| 00DD | E206 30C1 | STR D,6,0+CNDX | 353 |
| 00DF | 6056 | LDR IS,6,5 | 354 |
| 00E0 | E207 30BF | ADD D,7,0+PRI | 355 |
| 00E2 | 9B67 30C2 | STR DX,7,0+CHAN,6 | 356 |
| | | FLAG PULSE TRAIN AS STABLE AND INCREMENT CHANNEL COUNTER "/ | 357 |
| | | SSF(,@I/5,)=TRUE | 358 |
| 00E4 | E0A8 | LDR R,8,10 | 359 |
| 00E5 | D8F7 | RA,7,16 | 360 |
| 00E6 | 4052 | IS,2,5 | 361 |
| 00E7 | FC27 | R,2,2 | 362 |
| 00E8 | 4012 | IS,2,1 | 363 |
| 00E9 | 9B82 3163 | STR DX,2,0+SSF,8 | 364 |
| | | NC=@NC+1 | 365 |
| 00EB | E202 30B8 | LDR D,2,0+NC | 366 |
| 00ED | 6012 | IS,2,1 | 367 |
| 00EE | 9C02 30B8 | STR D,2,0+NC | 368 |
| | | RETURN | 369 |
| 00F0 | BF17 | R,7,1 | 370 |
| | | BFC | 371 |
| | | | 372 |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|-----|-------------|---|---------|
| | | ./" INITIALIZE VARIABLES "/ | |
| | | : CNDX=-6 | 426 |
| | | ORG RQM | 427 |
| | | EQU \$ | 428 |
| | | LDR IS,12,0 | 429 |
| | | LDR IS,2,-6 | 430 |
| | | STR D,2,0+CNDX | 431 |
| | | SCANTM=1120 | 432 |
| | | LDR I,2,1120 | 433 |
| | | STR D,2,0+SCANTM | 434 |
| | | I=0 | 435 |
| | | LDR R,10,12 | 436 |
| | | BP=0 | 437 |
| | | LDR R,11,12 | 438 |
| | | NC=1 | 439 |
| | | LDR IS,2,1 | 440 |
| | | STR D,2,0+NC | 441 |
| | | ./" SET NEW SIGNAL POINTER FILE TO NEGATIVE VALUES "/ | 442 |
| | | : ITERATE 200 TIMES | 443 |
| | | NS(.a1.)=-1 | 444 |
| | | LDR I,2,-201 | 445 |
| | | BRC I,7,a22 | 446 |
| | | EQU \$ | 447 |
| | | LDR R,3,10 | 448 |
| | | LDR IS,4,-1 | 449 |
| | | STR DX,4,0+NS,3 | 450 |
| | | I=a1+1 | 451 |
| | | LDR R,4,10 | 452 |
| | | ADD IS,4,1 | 453 |
| | | LDR R,10,4 | 454 |
| | | END | 455 |
| | | a22 EQU \$ | 456 |
| | | I=0 | 457 |
| | | LDR R,10,12 | 458 |
| | | NS(.a1.)=0 | 459 |
| | | LDR R,4,10 | 460 |
| | | STR DX,12,0+NS,4 | 461 |
| | | ./" LOAD INPUT BUFFER FROM CARDS "/ | 462 |
| | | : READ_DEC(BUF,BUF+3999) | 463 |
| | | SIM I,a23 | 464 |
| | | BRC IS,7,a24 | 465 |
| | | EQU \$ | 466 |
| | | DC O+BUF | 467 |
| | | DC 3999+BUF | 468 |
| | | EQU \$ | 469 |
| | | a23 | 470 |
| | | a24 | 471 |
| | | | 472 |
| | | | 473 |
| | | | 474 |
| | | | 475 |
| | | | 476 |
| | | | 477 |
| | | | 478 |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|--|---------|
| 011C | E0B4 | MAINLP: | 479 |
| 011D | 6024 | DO EQU \$ | 480 |
| 011E | E345 | 225 | 481 |
| 0120 | 9C05 | 20C0 31E7 | 482 |
| 0122 | E0B5 | LOAD OP TOA OF FIRST PULSE OF SIGNAL FROM INPUT BUFFER AND | 483 |
| 0123 | 6035 | COMPUTE EXPECTED TIME OF ARRIVAL OF LAST PULSE BASED ON RCVR | 484 |
| 0124 | E356 | ANTENNA SCAN TIME FOR ONE ANTENNA BEAMWIDTH "/ | 485 |
| 0126 | 9C06 | 2000 31E8 | 486 |
| 0128 | BC1E | TOAL=2BUF(.1@BP+2).) | 487 |
| 012A | A90E | LDR R,4,11 | 488 |
| 012C | 9C1E | ADD IS,4,2 | 489 |
| | | LDR DX,5,0+BUF,4 | 490 |
| | | STR D,5,0+TOAL | 491 |
| | | TOAU=2BUF(.1@BP+3).) | 492 |
| | | LDR R,5,11 | 493 |
| | | ADD IS,5,3 | 494 |
| | | LDR DX,6,0+BUF,5 | 495 |
| | | STR D,6,0+TOAU | 496 |
| | | CODE | 497 |
| | | LORM D,14,TOAL,2 | 498 |
| | | DADD D,14,SCANTM | 499 |
| | | STRM D,14,ITEM,2 | 500 |
| | | END | 501 |
| | | STR=2I/5*2 | 502 |
| 012E | E0A7 | R,7,10 | 503 |
| 012F | D8F6 | RA,6,16 | 504 |
| 0130 | 4058 | IS,8,5 | 505 |
| 0131 | FC86 | R,6,8 | 506 |
| 0132 | CD07 | I,7,2 | 507 |
| 0134 | 5C07 | STR D,7,0+STR | 508 |
| | | ITEM | 509 |
| 0136 | E208 | LDR D,8,0+STR | 510 |
| 0138 | E202 | LDR D,2,0+ITEM | 511 |
| 013A | 9B82 | STR DX,2,0+TMAX,8 | 512 |
| | | TMAX(.1@STR+1).)=@ITEM(.1.) | 513 |
| | | LDR D,2,0+STR | 514 |
| 013C | E202 | LDR D,2,0+STR | 515 |
| 013E | 6012 | ADD IS,2,1 | 516 |
| 013F | E203 | LDR D,3,1+ITEM | 517 |
| 0141 | 9B23 | STR DX,3,0+TMAX,2 | 518 |
| | | TMIN(.1@STR+1).)=@TOAL | 519 |
| 0143 | E203 | LDR D,3,0+STR | 520 |
| 0145 | E204 | LDR D,4,0+TOAL | 521 |
| 0147 | 9B34 | STR DX,4,0+TMIN,3 | 522 |
| | | TMIN(.1@STR+1).)=@TOAU | 523 |
| 0149 | E204 | LDR D,4,0+STR | 524 |
| 014B | 6014 | ADD IS,4,1 | 525 |
| 014C | E205 | LDR D,5,0+TOAU | 526 |
| 014E | 9B45 | STR DX,5,0+TMIN,4 | 527 |
| | | COMPUTE: | 528 |
| | | DO EQU \$ | 529 |
| | | 227 | 530 |
| | | "/" INCREMENT BUFFER POINTER AND CHECK AT BUFFER LIMIT. IFSC, GO | 531 |

| LCC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|--|---------|
| | | COMPRESS RESIDUAL PULSES WITHIN INPUT BUFFER FOR FUTHER PROCESSING | |
| 0150 | E0B5 | BP=BP+5 | 532 |
| 0151 | E055 | LDR R,5,11 | 533 |
| 0152 | E058 | ADD R,5,5 | 534 |
| 0153 | E0B5 | LDR R,11,5 | 535 |
| 0154 | B105 0F9F | TEST BP>3999 | 536 |
| | | LDR R,5,11 | 537 |
| | | CMP I,5,3999 | 538 |
| 0150 | C106 015A | IFSO: LEAVE MAINLP | 539 |
| 0158 | C107 02CD | EQU \$ | 540 |
| | | BRC I,6,032 | 541 |
| | | END I,7,026 | 542 |
| | | EQU \$ | 543 |
| | | EQU \$ | 544 |
| | | IF NO SIGNAL IN BUFFER INDEXED BY BUFFER POINTER INCREMENT | 545 |
| | | POINTER AND REPEAT COMPUTE "/ | 546 |
| | | TEST BPBP(.BPBP.) | 547 |
| 015A | E0B5 | LDR R,5,11 | 548 |
| 015B | E356 2000 | LDR DX,6,0+BUF,5 | 549 |
| | | ZERO: REPEAT COMPUTE | 550 |
| 015D | C105 0161 | EQU \$ | 551 |
| 015F | C107 0150 | BRC I,5,036 | 552 |
| | | END I,7,027 | 553 |
| | | EQU \$ | 554 |
| | | EQU \$ | 555 |
| | | I=0 | 556 |
| 0161 | E0CA | LDR R,10,12 | 557 |
| 0162 | E0CS | LDR R,9,12 | 558 |
| | | SORT: | 559 |
| | | DO | 560 |
| | | IF AT END OF NEW SIGNAL POINTER FILE, LEAVE SORT "/ | 561 |
| | | TEST AI>195 | 562 |
| 0163 | E0A6 | LDR R,6,10 | 563 |
| 0164 | B106 00C3 | CMP I,6,195 | 564 |
| | | IFSO: LEAVE SORT | 565 |
| | | EQU \$ | 566 |
| 0166 | C106 016A | BRC I,6,042 | 567 |
| 0168 | C107 02B3 | END I,7,038 | 568 |
| | | EQU \$ | 569 |
| | | EQU \$ | 570 |
| | | IF NEW SIGNAL POINTER FILE INDEXED BY I IS EMPTY INDEX AND | 571 |
| | | "/ | 572 |
| | | END | 573 |
| | | EQU \$ | 574 |
| | | EQU \$ | 575 |
| | | IF NEW SIGNAL POINTER FILE INDEXED BY I IS EMPTY INDEX AND | 576 |
| | | "/ | 577 |
| | | END | 578 |
| | | EQU \$ | 579 |
| | | EQU \$ | 580 |
| | | IF NEW SIGNAL POINTER FILE INDEXED BY I IS EMPTY INDEX AND | 581 |
| | | "/ | 582 |
| | | END | 583 |
| | | EQU \$ | 584 |
| | | EQU \$ | 585 |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|------------|---------|
| 016A | E0A6 | • | 585 |
| 016B | E367 2FA0 | • | 586 |
| 016D | C103 0174 | • | 587 |
| 016F | E0A7 | • | 588 |
| 0170 | E057 | • | 589 |
| 0171 | E07A | • | 590 |
| 0172 | C107 0163 | • | 591 |
| 0174 | E0B8 | • | 592 |
| 0175 | 6C27 | • | 593 |
| 0176 | E378 2000 | • | 594 |
| 0177 | 9C08 323D | • | 595 |
| 017A | E0B8 | • | 596 |
| 017B | 6038 | • | 597 |
| 017C | E382 2000 | • | 598 |
| 017E | 9C02 323E | • | 599 |
| 0180 | E0A3 | • | 600 |
| 0181 | D8F2 | • | 601 |
| 0182 | 4054 | • | 602 |
| 0183 | FC42 | • | 603 |
| 0184 | C003 | • | 604 |
| 0186 | E334 | • | 605 |
| 0188 | 9C04 323F | • | 606 |
| 018A | E0A5 | • | 607 |
| 018B | D8F4 | • | 608 |
| 018C | 4056 | • | 609 |
| 018D | FC54 | • | 610 |
| 018E | C005 | • | 611 |
| 0190 | 6015 | • | 612 |
| 0191 | E356 31E9 | • | 613 |
| 0192 | E0A5 | • | 614 |
| 0193 | D8F4 | • | 615 |
| 0194 | 4056 | • | 616 |
| 0195 | FC54 | • | 617 |
| 0196 | C005 | • | 618 |
| 0197 | 6015 | • | 619 |
| 0198 | E356 31E9 | • | 620 |
| 0199 | E0A5 | • | 621 |
| 0200 | D8F4 | • | 622 |
| 0201 | 4056 | • | 623 |
| 0202 | FC54 | • | 624 |
| 0203 | C005 | • | 625 |
| 0204 | 6015 | • | 626 |
| 0205 | E356 31E9 | • | 627 |
| 0206 | E0A5 | • | 628 |
| 0207 | D8F4 | • | 629 |
| 0208 | 4056 | • | 630 |
| 0209 | FC54 | • | 631 |
| 0210 | C005 | • | 632 |
| 0211 | 6015 | • | 633 |
| 0212 | E356 31E9 | • | 634 |
| 0213 | E0A5 | • | 635 |
| 0214 | D8F4 | • | 636 |
| 0215 | 4056 | • | 637 |
| 0216 | FC54 | • | 638 |
| 0217 | C005 | • | 639 |
| 0218 | 6015 | • | 640 |
| 0219 | E356 31E9 | • | 641 |
| 0220 | E0A5 | • | 642 |
| 0221 | D8F4 | • | 643 |
| 0222 | 4056 | • | 644 |
| 0223 | FC54 | • | 645 |
| 0224 | C005 | • | 646 |
| 0225 | 6015 | • | 647 |
| 0226 | E356 31E9 | • | 648 |
| 0227 | E0A5 | • | 649 |
| 0228 | D8F4 | • | 650 |
| 0229 | 4056 | • | 651 |
| 0230 | FC54 | • | 652 |
| 0231 | C005 | • | 653 |
| 0232 | 6015 | • | 654 |
| 0233 | E356 31E9 | • | 655 |
| 0234 | E0A5 | • | 656 |
| 0235 | D8F4 | • | 657 |
| 0236 | 4056 | • | 658 |
| 0237 | FC54 | • | 659 |
| 0238 | C005 | • | 660 |
| 0239 | 6015 | • | 661 |
| 0240 | E356 31E9 | • | 662 |
| 0241 | E0A5 | • | 663 |
| 0242 | D8F4 | • | 664 |
| 0243 | 4056 | • | 665 |
| 0244 | FC54 | • | 666 |
| 0245 | C005 | • | 667 |
| 0246 | 6015 | • | 668 |
| 0247 | E356 31E9 | • | 669 |
| 0248 | E0A5 | • | 670 |
| 0249 | D8F4 | • | 671 |
| 0250 | 4056 | • | 672 |
| 0251 | FC54 | • | 673 |
| 0252 | C005 | • | 674 |
| 0253 | 6015 | • | 675 |
| 0254 | E356 31E9 | • | 676 |
| 0255 | E0A5 | • | 677 |
| 0256 | D8F4 | • | 678 |
| 0257 | 4056 | • | 679 |
| 0258 | FC54 | • | 680 |
| 0259 | C005 | • | 681 |
| 0260 | 6015 | • | 682 |
| 0261 | E356 31E9 | • | 683 |
| 0262 | E0A5 | • | 684 |
| 0263 | D8F4 | • | 685 |
| 0264 | 4056 | • | 686 |
| 0265 | FC54 | • | 687 |
| 0266 | C005 | • | 688 |
| 0267 | 6015 | • | 689 |
| 0268 | E356 31E9 | • | 690 |
| 0269 | E0A5 | • | 691 |
| 0270 | D8F4 | • | 692 |
| 0271 | 4056 | • | 693 |
| 0272 | FC54 | • | 694 |
| 0273 | C005 | • | 695 |
| 0274 | 6015 | • | 696 |
| 0275 | E356 31E9 | • | 697 |
| 0276 | E0A5 | • | 698 |
| 0277 | D8F4 | • | 699 |
| 0278 | 4056 | • | 700 |
| 0279 | FC54 | • | 701 |
| 0280 | C005 | • | 702 |
| 0281 | 6015 | • | 703 |
| 0282 | E356 31E9 | • | 704 |
| 0283 | E0A5 | • | 705 |
| 0284 | D8F4 | • | 706 |
| 0285 | 4056 | • | 707 |
| 0286 | FC54 | • | 708 |
| 0287 | C005 | • | 709 |
| 0288 | 6015 | • | 710 |
| 0289 | E356 31E9 | • | 711 |
| 0290 | E0A5 | • | 712 |
| 0291 | D8F4 | • | 713 |
| 0292 | 4056 | • | 714 |
| 0293 | FC54 | • | 715 |
| 0294 | C005 | • | 716 |
| 0295 | 6015 | • | 717 |
| 0296 | E356 31E9 | • | 718 |
| 0297 | E0A5 | • | 719 |
| 0298 | D8F4 | • | 720 |
| 0299 | 4056 | • | 721 |
| 0300 | FC54 | • | 722 |
| 0301 | C005 | • | 723 |
| 0302 | 6015 | • | 724 |
| 0303 | E356 31E9 | • | 725 |
| 0304 | E0A5 | • | 726 |
| 0305 | D8F4 | • | 727 |
| 0306 | 4056 | • | 728 |
| 0307 | FC54 | • | 729 |
| 0308 | C005 | • | 730 |
| 0309 | 6015 | • | 731 |
| 0310 | E356 31E9 | • | 732 |
| 0311 | E0A5 | • | 733 |
| 0312 | D8F4 | • | 734 |
| 0313 | 4056 | • | 735 |
| 0314 | FC54 | • | 736 |
| 0315 | C005 | • | 737 |
| 0316 | 6015 | • | 738 |
| 0317 | E356 31E9 | • | 739 |
| 0318 | E0A5 | • | 740 |
| 0319 | D8F4 | • | 741 |
| 0320 | 4056 | • | 742 |
| 0321 | FC54 | • | 743 |
| 0322 | C005 | • | 744 |
| 0323 | 6015 | • | 745 |
| 0324 | E356 31E9 | • | 746 |
| 0325 | E0A5 | • | 747 |
| 0326 | D8F4 | • | 748 |
| 0327 | 4056 | • | 749 |
| 0328 | FC54 | • | 750 |
| 0329 | C005 | • | 751 |
| 0330 | 6015 | • | 752 |
| 0331 | E356 31E9 | • | 753 |
| 0332 | E0A5 | • | 754 |
| 0333 | D8F4 | • | 755 |
| 0334 | 4056 | • | 756 |
| 0335 | FC54 | • | 757 |
| 0336 | C005 | • | 758 |
| 0337 | 6015 | • | 759 |
| 0338 | E356 31E9 | • | 760 |
| 0339 | E0A5 | • | 761 |
| 0340 | D8F4 | • | 762 |
| 0341 | 4056 | • | 763 |
| 0342 | FC54 | • | 764 |
| 0343 | C005 | • | 765 |
| 0344 | 6015 | • | 766 |
| 0345 | E356 31E9 | • | 767 |
| 0346 | E0A5 | • | 768 |
| 0347 | D8F4 | • | 769 |
| 0348 | 4056 | • | 770 |
| 0349 | FC54 | • | 771 |
| 0350 | C005 | • | 772 |
| 0351 | 6015 | • | 773 |
| 0352 | E356 31E9 | • | 774 |
| 0353 | E0A5 | • | 775 |
| 0354 | D8F4 | • | 776 |
| 0355 | 4056 | • | 777 |
| 0356 | FC54 | • | 778 |
| 0357 | C005 | • | 779 |
| 0358 | 6015 | • | 780 |
| 0359 | E356 31E9 | • | 781 |
| 0360 | E0A5 | • | 782 |
| 0361 | D8F4 | • | 783 |
| 0362 | 4056 | • | 784 |
| 0363 | FC54 | • | 785 |
| 0364 | C005 | • | 786 |
| 0365 | 6015 | • | 787 |
| 0366 | E356 31E9 | • | 788 |
| 0367 | E0A5 | • | 789 |
| 0368 | D8F4 | • | 790 |
| 0369 | 4056 | • | 791 |
| 0370 | FC54 | • | 792 |
| 0371 | C005 | • | 793 |
| 0372 | 6015 | • | 794 |
| 0373 | E356 31E9 | • | 795 |
| 0374 | E0A5 | • | 796 |
| 0375 | D8F4 | • | 797 |
| 0376 | 4056 | • | 798 |
| 0377 | FC54 | • | 799 |
| 0378 | C005 | • | 800 |
| 0379 | 6015 | • | 801 |
| 0380 | E356 31E9 | • | 802 |
| 0381 | E0A5 | • | 803 |
| 0382 | D8F4 | • | 804 |
| 0383 | 4056 | • | 805 |
| 0384 | FC54 | • | 806 |
| 0385 | C005 | • | 807 |
| 0386 | 6015 | • | 808 |
| 0387 | E356 31E9 | • | 809 |
| 0388 | E0A5 | • | 810 |
| 0389 | D8F4 | • | 811 |
| 0390 | 4056 | • | 812 |
| 0391 | FC54 | • | 813 |
| 0392 | C005 | • | 814 |
| 0393 | 6015 | • | 815 |
| 0394 | E356 31E9 | • | 816 |
| 0395 | E0A5 | • | 817 |
| 0396 | D8F4 | • | 818 |
| 0397 | 4056 | • | 819 |
| 0398 | FC54 | • | 820 |
| 0399 | C005 | • | 821 |
| 0400 | 6015 | • | 822 |
| 0401 | E356 31E9 | • | 823 |
| 0402 | E0A5 | • | 824 |
| 0403 | D8F4 | • | 825 |
| 0404 | 4056 | • | 826 |
| 0405 | FC54 | • | 827 |
| 0406 | C005 | • | 828 |
| 0407 | 6015 | • | 829 |
| 0408 | E356 31E9 | • | 830 |
| 0409 | E0A5 | • | 831 |
| 0410 | D8F4 | • | 832 |
| 0411 | 4056 | • | 833 |
| 0412 | FC54 | • | 834 |
| 0413 | C005 | • | 835 |
| 0414 | 6015 | • | 836 |
| 0415 | E356 31E9 | • | 837 |
| 0416 | E0A5 | • | 838 |
| 0417 | D8F4 | • | 839 |
| 0418 | 4056 | • | 840 |
| 0419 | FC54 | • | 841 |
| 0420 | C005 | • | 842 |
| 0421 | 6015 | • | 843 |
| 0422 | E356 31E9 | • | 844 |
| 0423 | E0A5 | • | 845 |
| 0424 | D8F4 | • | 846 |
| 0425 | 4056 | • | 847 |
| 0426 | FC54 | • | 848 |
| 0427 | C005 | • | 849 |
| 0428 | 6015 | • | 850 |
| 0429 | E356 31E9 | • | 851 |
| 0430 | E0A5 | • | 852 |
| 0431 | D8F4 | • | 853 |
| 0432 | 4056 | • | 854 |
| 0433 | FC54 | • | 855 |
| 0434 | C005 | • | 856 |
| 0435 | 6015 | • | 857 |
| 0436 | E356 31E9 | • | 858 |
| 0437 | E0A5 | • | 859 |
| 0438 | D8F4 | • | 860 |
| 0439 | 4056 | • | 861 |
| 0440 | FC54 | • | 862 |
| 0441 | C005 | • | 863 |
| 0442 | 6015 | • | 864 |
| 0443 | E356 31E9 | • | 865 |
| 0444 | E0A5 | • | 866 |
| 0445 | D8F4 | • | 867 |
| 0 | | | |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|---------------------------------|---------|
| 0153 | 9C06 3240 | STR D,6,1+CON2 | 638 |
| 0195 | BC1E 323D | LDRM D,14,CON1,2 | 639 |
| 0197 | BC17 323F | LDRM D,7,CON2,2 | 640 |
| 0199 | AAE7 | DSUB R,7,14 | 641 |
| 019A | 9C17 3239 | STRM D,7,1TEST,2 | 642 |
| 019C | E206 3239 | END TEST | 643 |
| 019E | E206 3239 | TEST | 644 |
| 019F | E206 3239 | TEST | 645 |
| 01A0 | E0A7 | NEG,ZERO: DO | 646 |
| 01A1 | D8F6 | EQU \$ | 647 |
| 01A2 | 4058 | BRC I,1,250 | 648 |
| 01A3 | FC86 | TEST @SSF(,(@I/5).) | 649 |
| 01A4 | E378 3163 | LDR R,7,10 | 650 |
| 01A6 | C106 01EE | SHD RA,6,16 | 651 |
| 01A8 | E208 30B8 | LDR IS,8,5 | 652 |
| 01AA | GFF8 | DIV R,6,8 | 653 |
| 01AB | 9C08 30B8 | LDR DX,8,0+SSF,7 | 654 |
| 01AD | E0A3 | PGS: DO | 655 |
| 01AE | D8F2 | EQU \$ | 656 |
| 01AF | 4054 | BRC I,6,254 | 657 |
| 01B0 | FC42 | LDK D,8,0+NC | 658 |
| 01B1 | 983C 3163 | ADD IS,8,-1 | 659 |
| 01B3 | E0A4 | STR C,8,0+NC | 660 |
| 01B4 | E345 | SSF(,(@I/5).)=FALSE | 661 |
| 01B5 | E356 2C00 | LDR R,3,10 | 662 |
| 01B8 | 9C06 31B9 | SHD RA,2,16 | 663 |
| 01BA | 9C0C 31E4 | LDR IS,4,5 | 664 |
| 01BC | 4FA6 | DIV R,2,4 | 665 |
| 01BD | C107 01D9 | STR DX,12,0+SSF,3 | 666 |
| 01BF | 9C0C 31E5 | LDR R,4,10 | 667 |
| 01C1 | 4FA7 | LDR DX,5,0+NS,4 | 668 |
| 01C2 | C107 01D2 | LDR DX,6,0+BUF,5 | 669 |
| 01C4 | E0A8 | STR D,6,0+BRNGF | 670 |
| 01C5 | 82C8 31E4 | I=0 | 671 |
| 01C7 | E362 2FA0 | ITERATE 5 TIMES | 672 |
| 01C9 | 8202 31E5 | STR D,12,0+II | 673 |
| | | ITERATE 5 TIMES | 674 |
| | | JJ=0 | 675 |
| | | LDR IS,6,-6 | 676 |
| | | BRC I,7,256 | 677 |
| | | EQU \$ | 678 |
| | | STR D,12,0+JJ | 679 |
| | | ITERATE 5 TIMES | 680 |
| | | BUF(,(@NSI(,(@I+@II).)+@JJ).)=0 | 681 |
| | | I,7,258 | 682 |
| | | LDR IS,7,-6 | 683 |
| | | BRC I,7,258 | 684 |
| | | EQU \$ | 685 |
| | | LDR R,8,10 | 686 |
| | | LDR D,8,0+II | 687 |
| | | ADD DX,2,0+NS,8 | 688 |
| | | LDR D,2,0+JJ | 689 |
| | | ADD | 690 |

| LUC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|--|---------|
| 01C6 | 9B2C 2000 | STR DX,12,0+BUF,2 JJ=@JJ+1 | 691 |
| 01CD | E202 31E5 | LDR D,2,0+JJ | 692 |
| 01CF | 6012 | ADD IS,2,1 | 693 |
| 01D0 | 9C02 31E5. | STR D,2,0+JJ END | 694 |
| | | | 695 |
| | | | 696 |
| 01D2 | C907 01C4 | EQU \$ I,7,@57 IBN II=@II+1 | 697 |
| | | | 698 |
| 01D4 | E202 31E4 | LDR D,2,0+II | 699 |
| 01D6 | 6012 | ADD IS,2,1 | 700 |
| 01D7 | 9C02 31E4 | STR D,2,0+II END | 701 |
| | | | 702 |
| | | | 703 |
| 01D9 | C906 01BF | EQU \$ I,6,@55 IBN CHAN(.@CINDX(.@I/5.).)=(@BRNGF+@BRNGL(.@I/5.))/2 | 704 |
| | | | 705 |
| 01CB | EOA3 | LDR R,3,10 | 706 |
| 01CC | D8F2 | SHD RA,2,16 | 707 |
| 01CD | 4054 | LDR IS,4,5 | 708 |
| 01DE | FC42 | DIV R,2,4 | 709 |
| 01DF | E334 31B8 | LDR DX,4,0+CINDX,3 | 710 |
| 01E1 | EOA6 | LDR RA,6,10 | 711 |
| 01E2 | D8F5 | SHD RA,5,16 | 712 |
| 01E3 | 4057 | LDR IS,7,5 | 713 |
| 01E4 | FC75 | DIV R,5,7 | 714 |
| 01E5 | E207 31B9 | LDR D,7,0+BRNGF | 715 |
| 01E7 | 8367 31BA | ADD DX,7,0+BRNGL,6 | 716 |
| 01E9 | D8F6 | SHD RA,8,16 | 717 |
| 01EA | 4028 | LDR IS,8,2 | 718 |
| 01EB | FC06 | DIV R,6,8 | 719 |
| 01EC | 9B47 30C2 | STR DX,7,0+CHAN,4 END | 720 |
| | | | 721 |
| | | | 722 |
| | | | 723 |
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| | | | 743 |

```

. @54 $ $
. @51 $ $
. /" IF NS POINTER FILE SIGNAL HAS BEEN CLASSIFIED AS NON STABLE
. RESET NSSF FLAG "/"
. NSSF(.@I/5.)=FALSE
. LDR R,3,10
. SHD RA,2,16
. LDR IS,4,5
. DIV R,2,4
. STR DX,12,0+NSSF,3
. /" CLEAR THE SIGNAL FROM NS POINTER FILE "/"
. II=0
. STR D,12,0+II
. ITERATE 5 TIMES
. NS(.@I+@II.)=-1
. LDR IS,4,-6

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| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|---------------------------|---------|
| 021D | 6042 | ADD IS,2,4 | 797 |
| 021E | E0A3 | LDR R,3,10 | 798 |
| 021F | E334 | LDR DX,4,0+NS,3 | 799 |
| 0221 | 6044 | ADD IS,4,4 | 800 |
| 0222 | E325 | LDR DX,5,0+BUF,2 | 801 |
| 0224 | B343 | CMR DX,5,0+BUF,4 | 802 |
| | | IF NOT: DQ | 803 |
| | | EQU \$ | 804 |
| | | BRC I,2,@68 | 805 |
| 0226 | C102 022D | I=@1+5 | 806 |
| 0228 | E0A5 | LDR R,5,10 | 807 |
| 0229 | 6055 | ADD IS,5,5 | 808 |
| 022A | E05A | LDR R,10,5 | 809 |
| | | REPEAT SORT | 810 |
| 022B | C107 0163 | BRC I,7,@37 | 811 |
| | | END | 812 |
| | | END | 813 |
| | | EQU \$ | 814 |
| | | EQU \$ | 815 |
| | | TEST @SSF(,@1/5).) | 816 |
| 022D | E0A6 | LDR R,6,10 | 817 |
| 022E | D8F5 | SHD RA,5,16 | 818 |
| 022F | 4057 | LDR IS,7,5 | 819 |
| 0230 | FC75 | DIV R,5,7 | 820 |
| 0231 | E367 3163 | LDR DX,7,0+SSF,6 | 821 |
| | | NCNZERO: DU | 822 |
| | | EQU \$ | 823 |
| 0233 | C102 C251 | BRC I,2,@72 | 824 |
| | | BRNGL(,@1/5.)=@BUF(,@BP.) | 825 |
| 0235 | E0A8 | LDR R,8,10 | 826 |
| 0236 | D8F7 | SHD RA,7,16 | 827 |
| 0237 | 4052 | LDR IS,2,5 | 828 |
| 0238 | FC27 | DIV R,7,2 | 829 |
| 0239 | E0B2 | LDR R,2,11 | 830 |
| 023A | E323 2000 | LDR DX,3,0+BUF,2 | 831 |
| 023C | 9B33 31BA | STR DX,3,0+BRNGL,8 | 832 |
| | | I1=0 | 833 |
| 023E | 9C0C 31E4 | STR D,12,0+I1 | 834 |
| | | ITERATE 5 TIMES | 835 |
| | | BUF(,@BP+@I1.)=0 | 836 |
| 0240 | 4FA3 | LDR IS,3,-6 | 837 |
| 0241 | C1C7 0240 | I,7,@74 | 838 |
| | | \$ | 839 |
| 0243 | E0B4 | R,4,11 | 840 |
| 0244 | E204 31E4 | D,4,0+I1 | 841 |
| 0246 | 9B4C 2000 | STR DX,12,0+BUF,4 | 842 |
| | | I1=@I1+1 | 843 |
| 0248 | E204 31E4 | LDR D,4,0+I1 | 844 |
| 024A | 6014 | ADD IS,4,1 | 845 |
| 024B | 5C04 31E4 | STR D,4,0+I1 | 846 |
| | | END | 847 |
| | | EQU \$ | 848 |
| 024D | C903 0243 | I,3,@73 | 849 |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|---|---|
| 024F | C107 0150 | REPEAT COMPUTE BRC I,7,227 END @72 @69 END TEST JNSSF(, (21/5),) == TRUE LDR RA,4,15 SHD IS,6,5 DIV R,4,6 LDR DX,6,0+NSSF,5 CMP IS,6,1 IFSU: REPEAT COMPUTE @77 END @78 @75 END @79 END @83 END @84 @81 @80 | 850 851 852 853 854 855 856 857 858 859 860 861 862 863 864 865 866 867 868 869 870 871 872 873 874 875 876 877 878 879 880 881 882 883 884 885 886 887 888 889 890 891 892 893 894 895 896 897 898 899 900 901 902 |
| 0251 | E0A5 | | |
| 0252 | D8F4 | | |
| 0253 | 4056 | | |
| 0254 | FC64 | | |
| 0255 | E356 | | |
| 0257 | 2016 | | |
| 0258 | C105 025C | | |
| 025A | C107 0150 | | |
| 025C | 4FC6 | | |
| 025D | C107 026E | | |
| 025F | E057 | | |
| 0260 | 6017 | | |
| 0261 | E079 | | |
| 0262 | E0A7 | | |
| 0263 | 8097 | | |
| 0264 | E378 2FA0 | | |
| 0266 | C103 026E | | |
| 0268 | E0A8 | | |
| 0269 | 8098 | | |
| 026A | 9B8B 2FA0 | | |
| 026C | C107 0150 | | |

IF SIGNAL HAS NOT BEEN CLASSIFIED, STORE BUFFER POINTER IN
NS POINTER FILE. IF THIS IS THE FIFTH PULSE ASSOCIATED WITH
THIS TRAIN, COMPUTE TIME INTERVALS BETWEEN PULSES AND CALL
FOR FBI COMPUTATION.

| LUC | OBJECT CODE | CARD | IMAGE | CARDNUM |
|------|-------------|--|--------------------------|---------|
| 026E | C906 025F | 18N | I,6,a79 | 903 |
| 0270 | E0A8 | NS(. (a1+4) .)=aBP | R,6,10 | 904 |
| 0271 | 6048 | LDR | IS,8,10 | 905 |
| 0272 | 9B8B 2FA0 | ADD | IS,8,4 | 906 |
| 0274 | E208 30B8 | STR | DX,11,0+NS,8 | 907 |
| 0276 | 20A8 | TEST aNC==10 | LDR | 908 |
| | | LDR | D,8,0+NC | 909 |
| | | CMP | IS,8,10 | 910 |
| | | IFSU: DO | | 911 |
| 0277 | C105 0282 | EQU | I,5,a88 | 912 |
| 0279 | E0A3 | BRC | NSSF(. (a1/5) .)=TRUE | 913 |
| 027A | 0BF2 | LDR | R,3,10 | 914 |
| 027B | 4054 | SHD | RA,2,16 | 915 |
| 027C | FC42 | LDR | IS,4,5 | 916 |
| 027D | 4014 | OIV | R,2,4 | 917 |
| 027E | 5B34 313B | LDR | IS,4,1 | 918 |
| 0280 | C107 0150 | STR | DX,4,0+NSSF,3 | 919 |
| | | REPEAT COMPUTE | | 920 |
| | | BRC | 1,7,a27 | 921 |
| | | END | | 922 |
| | | END | | 923 |
| | | EQU | \$ | 924 |
| | | EQU | \$ | 925 |
| 0282 | 4019 | J=1 | | 926 |
| 0283 | 4FB4 | ITERATE 4 TIMES | IS,9,1 | 927 |
| 0284 | C107 02A8 | PI(.aJ.)=aBUF(. (aNS(. (a1+aJ) .)+2) .)-aBUF(. (aNS(. (a1+aJ-1) .)+2) .) | IS,4,-5 | 928 |
| | | LDR | 1,7,a90 | 929 |
| | | BRC | | 930 |
| | | EQU | \$ | 931 |
| | | LDR | R,5,9 | 932 |
| | | LDR | R,6,10 | 933 |
| | | ADD | R,6,9 | 934 |
| | | ADD | DX,7,0+NS,6 | 935 |
| | | ADD | IS,7,2 | 936 |
| | | ADD | R,8,10 | 937 |
| | | ADD | R,8,9 | 938 |
| | | ADD | IS,8,-1 | 939 |
| | | ADD | DX,2,0+NS,8 | 940 |
| | | ADD | IS,2,2 | 941 |
| | | ADD | DX,3,0+BUF,7 | 942 |
| | | SUB | DX,3,0+BUF,2 | 943 |
| | | STR | DX,3,0+PI,5 | 944 |
| | | TEST aPI(.aJ.) | | 945 |
| | | LDR | R,5,9 | 946 |
| | | LDR | DX,5,0+PI,3 | 947 |
| | | NEG: | PI(.aJ.)=aPI(.aJ.)+32767 | 948 |
| | | EQU | \$ | 949 |
| | | BRC | I,3,a94 | 950 |
| | | LDR | R,5,9 | 951 |
| | | LDR | R,0,9 | 952 |
| | | LDR | DX,7,0+PI,6 | 953 |
| | | LDR | | 954 |
| | | LDR | | 955 |

| LOC | OBJECT CODE | CARD | IMAGE | CARDNUM |
|------|-------------|----------|---------------------|---------|
| 02A1 | 8107 7FFF | | | 956 |
| 02A3 | 9857 30B9 | | ADD I,7,32767 | 957 |
| | | | STR DX,7,0+PI,5 | 958 |
| | | END | EQU \$ | 959 |
| | | END | EQU \$ | 960 |
| 02A5 | E097 | J=J+1 | LDR R,7,9 | 961 |
| 02A6 | 6017 | | ADD IS,7,1 | 962 |
| 02A7 | E079 | | LDR R,9,7 | 963 |
| | | END | | 964 |
| 02A8 | C904 0286 | | EQU \$ I,4,@89 | 965 |
| | | CALL | PRICOMP() | 966 |
| 02AA | 6010 | | ADD IS,0,1 | 967 |
| 02AB | 8DAC | | STK DR,0,11 | 968 |
| 02AC | ED01 00G0 | | PR I,1,0+PRICOMP | 969 |
| 02AE | 8EAO | | RET DR,0,11 | 970 |
| 02AF | 6FF0 | | ADD IS,0,-1 | 971 |
| 02B0 | 5C07 | | LDR RX,7,0,12 | 972 |
| | | REPEAT | COMPUTE I,7,@27 | 973 |
| 02B1 | C107 0150 | END SORT | EQU \$ | 974 |
| | | I=0 | LDR R,10,12 | 975 |
| 02B3 | E0CA | | CYCLE | 976 |
| | | TEST | ANS(.a1,) | 977 |
| 02B4 | E0A7 | | EQU \$ | 978 |
| 02B5 | E378 2FA0 | | LDR R,7,10 | 979 |
| | | NEG: DO | LDR DX,8,0+NS,7 | 980 |
| 02B7 | C103 02BE | | EQU \$ I,3,@99 | 981 |
| | | BRC | NS(.a1,)=@BP | 982 |
| 02B9 | E0A8 | | LDR R,8,10 | 983 |
| 02BA | 988B 2FA0 | | STR DX,11,0+NS,8 | 984 |
| 02BC | C107 011C | | REPEAT MAINLP | 985 |
| | | BRC | I,7,@25 | 986 |
| | | END | | 987 |
| 02BE | E0A8 | | EQU \$ | 988 |
| 02BF | 6058 | | EQU \$ I,3,@99 | 989 |
| 02C0 | E08A | | TEST(I=@I+5)>195 | 990 |
| 02C1 | B108 00C3 | | LDR R,8,10 | 991 |
| | | | ADD IS,8,5 | 992 |
| | | | LDR R,10,8 | 993 |
| | | | CMP I,8,195 | 994 |
| | | IFSO: DO | | 995 |
| 02C3 | C106 02CB | | EQU \$ | 996 |
| | | BRC | I,6,@103 | 997 |
| 02C5 | FE0D 02C8 | | DUMP DEC(NS,NS+195) | 998 |
| 02C7 | 0017 | | SIM I,3,@104 | 999 |
| | | BRC | IS,7,@105 | 1000 |
| | | | | 1001 |
| | | | | 1002 |
| | | | | 1003 |
| | | | | 1004 |
| | | | | 1005 |
| | | | | 1006 |
| | | | | 1007 |
| | | | | 1008 |

| LOC | CBJECT CODE | CARD IMAGE |
|------|-------------|--|
| 02C8 | 2FA0 | @104 EQU \$ Q*NS |
| 02C9 | 30C3 | DC 195+NS |
| 02CA | BF17 | @105 EQU \$ RETURN |
| | | END R,7,1 |
| | | END |
| | | @103 EQU \$ |
| | | @100 EQU \$ |
| | | END |
| 02C5 | C107 02B4 | BRC 1,7,@95 |
| | | END COMPUTE |
| | | @28 EQU \$ |
| | | END MAINLP |
| | | @20 EQU \$ |
| | | /" ROUTINE BELOW COMPRESSES RESIDUAL PULSES WITHIN THE INPUT |
| | | BUFFER. AN ATAC ASSEMBLY LANGUAGE PROGRAM TO PROCESS THIS |
| | | RESIDUAL MAY BE INCORPORATED BETWEEN THE END AND END MAIN |
| | | STATEMENTS UTILIZING THE CODE COLLECTION HEAD. "/ |
| | | BP=0 LDR R,11,i2 |
| 02CD | E0C6 | ZEROCK: CYCLE |
| | | % EQU \$ |
| | | @106 TEST @BUF(,@BP+1.) |
| | | @108 EQU \$ |
| | | LDR R,8,11 |
| | | ADD IS,8,1 |
| | | LDR DX,2,0+BUF,8 |
| | | ZERO: DC |
| | | @111 EQU \$ |
| | | BRC 1,5,@112 |
| | | OP=@BP |
| | | STR D,11,0+DP |
| | | LEAVE ZEROCK |
| | | BRC 1,7,@107 |
| | | END |
| | | @112 EQU \$ |
| | | @109 EQU \$ |
| | | BP=@BP+5 |
| | | LDR R,2,11 |
| | | ADD IS,2,5 |
| | | LDR R,11,2 |
| | | END ZEROCK |
| | | BRC 1,7,@108 |
| | | @107 EQU \$ |
| | | CYCLE |
| | | BP=@BP+5 |
| | | @113 EQU \$ |
| 02CE | E088 | |
| 02CF | 6018 | |
| 02D0 | E382 2000 | |
| 02D2 | C105 02D8 | |
| 02D4 | 9C08 31B3 | |
| 02D6 | C1C7 02DD | |
| 02L8 | E0E2 | |
| 02D9 | 6052 | |
| 02DA | E028 | |
| 02DB | C107 02CE | |

| LOC | OBJECT CODE | CARD | IMAGE | PAGE | CARDNUM |
|------|-------------|------|----------------------------------|------|---------|
| 02CD | E0B2 | | LDR R,2,11 | 1062 | |
| 02DE | 6052 | | ADD IS,2,5 | 1063 | |
| 02CF | E02B | | LDR R,11,2 | 1064 | |
| | | | TEST @BP>3995 | 1065 | |
| 02E0 | E0B2 | | LDR R,2,11 | 1066 | |
| 02E1 | B102 0F9B | | CMP I,2,3995 | 1067 | |
| | | | IFSO: DC | 1068 | |
| 02E3 | C106 02E6 | @116 | EQU \$ | 1069 | |
| | | | BRC I,6,@117 | 1070 | |
| 02E5 | BF17 | | RETURN | 1071 | |
| | | | BRC R,7,1 | 1072 | |
| | | | END | 1073 | |
| 02E6 | E0B2 | @117 | EQU \$ | 1074 | |
| 02E7 | 6012 | @114 | TEST @BUF(.(@BP+1).)=0 | 1075 | |
| 02E8 | E323 2000 | | LDR R,2,11 | 1076 | |
| 02EA | 2003 | | ADD IS,2,1 | 1077 | |
| | | | LDR DX,3,0+BUF,2 | 1078 | |
| | | | CMP IS,3,0 | 1079 | |
| | | | IFNOT: DO | 1080 | |
| 02EB | C102 0308 | @120 | EQU \$ | 1081 | |
| | | | BRC I,2,@121 | 1082 | |
| 02ED | E0CA | | I=0 | 1083 | |
| | | | LDR R,10,12 | 1084 | |
| | | | ITERATE 5 TIMES | 1085 | |
| | | | BUF(.(@DP+@I).)=@BUF(.(@BP+@I).) | 1086 | |
| 02EE | 4FA3 | | IS,3,-6 | 1087 | |
| 02EF | C107 0301 | @122 | I,7,@123 | 1088 | |
| | | | \$ | 1089 | |
| 02F1 | E204 31B3 | | D,4,0+DP | 1090 | |
| 02F3 | 80A4 | | R,4,10 | 1091 | |
| 02F4 | E0B5 | | R,5,11 | 1092 | |
| 02F5 | 80A5 | | R,5,10 | 1093 | |
| 02F6 | E356 2000 | | DX,6,0+BUF,5 | 1094 | |
| 02F8 | 9B46 2000 | | DX,6,0+BUF,4 | 1095 | |
| | | | BUF(.(@BP+@I).)=0 | 1096 | |
| 02FA | E0B6 | | R,6,11 | 1097 | |
| 02FB | 80A6 | | R,6,10 | 1098 | |
| 02FC | 9B6C 2000 | | DX,12,0+BUF,6 | 1099 | |
| | | | I=@I+1 | 1100 | |
| 02FE | E0A6 | | R,6,10 | 1101 | |
| 02FF | 6016 | | IS,6,1 | 1102 | |
| 0300 | E06A | | R,10,6 | 1103 | |
| | | @123 | END | 1104 | |
| 0301 | C903 02F1 | | \$ | 1105 | |
| | | | I,3,@122 | 1106 | |
| | | | DP=@DP+5 | 1107 | |
| 0303 | E206 31B3 | | D,6,0+DP | 1108 | |
| 0305 | 6050 | | IS,6,5 | 1109 | |
| 0306 | 5C06 31B3 | | D,6,0+DP | 1110 | |
| | | | END | 1111 | |
| | | | | 1112 | |
| | | | | 1113 | |
| | | | | 1114 | |

| LOC | OBJECT CODE | CARD IMAGE | CARDNUM |
|------|-------------|--|---------|
| | | @121 EQU \$ | 1115 |
| | | @118 EQU \$ | 1116 |
| 0308 | C107 02LD | . END BRC I,7,@113 | 1117 |
| 030A | BF17 | .END MAIN BRC R,7,1 | 1118 |
| | | . BRC R,7,1 | 1119 |
| | | :/***** LINK MODULE FOR EXECUTING PL/ATAC PROGRAM. | 1120 |
| | | .\$EXECUTIVE | 1121 |
| | | .\$***/ | 1122 |
| | | .\$ MODULE LINK | 1123 |
| | | ENTRY LINK | 1124 |
| | | DECLARE PROCSTK(64)RAM | 1125 |
| | | CRG RAM | 1126 |
| 3241 | | PROCSTK RS = PROCSTK | 1127 |
| | | CRG ROM | 1128 |
| | | LINK | 1129 |
| 0308 | 400C | EQU \$ | 1130 |
| 030C | E100 3241 | LDR I,12,0 | 1131 |
| | | LDR I,0,0+PROCSTK | 1132 |
| | | CALL MAIN() | 1133 |
| 030E | 601C | ADD IS,0,1 | 1134 |
| 030F | 8DA0 | STK DR,0,11 | 1135 |
| 0310 | EDC1 00FS | I,1,0+MAIN | 1136 |
| 0312 | 8EAC | BAL DR,0,11 | 1137 |
| 0313 | 6FFC | RET DR,0,11 | 1138 |
| 0314 | 5C02 | ADD IS,0,-1 | 1139 |
| | | LDR RX,2,0,12 | 1140 |
| 0315 | BF17 | END LINK R,7,1 | 1141 |
| | | .EOF EOF EOF EOF EOF EOF | 1142 |
| | | END | 1143 |
| | | | 1144 |
| | | | 1145 |
| | | | 1146 |

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BIBLIOGRAPHY

1. Frazer, P.D., The Electronic Warfare Application of Special Purpose Microprogrammed Minicomputers, M.S.E.E Thesis, Naval Postgraduate School, June, 1974.
2. ATAC, Applied Technology Airborne Computer, V. I,II, Itek Corporation, 1974.
3. Electronic Countermeasures, U.S. Army Signal Corps, 1961.
4. User's Manual, 1st ed., W.R. Church Computer Center, Naval Postgraduate School, Monterey, California, 1970.

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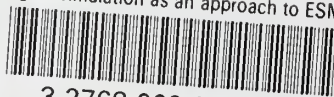
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